

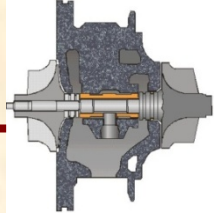
Vehicle Turbocharger Nonlinear Rotordynamics Modeling and Experimental Validation

Luis San Andrés

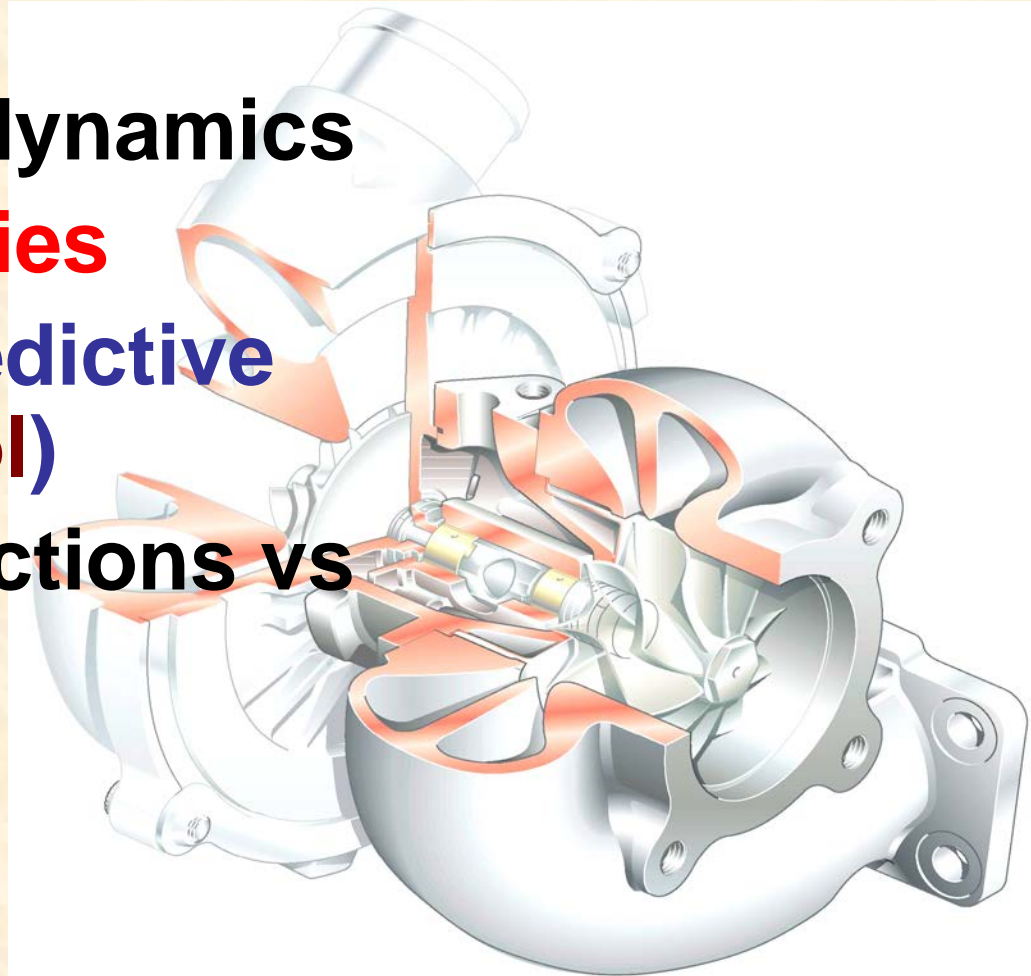
Mast-Childs Tribology Professor

Texas A&M University, Turbomachinery Laboratory

January, 2011

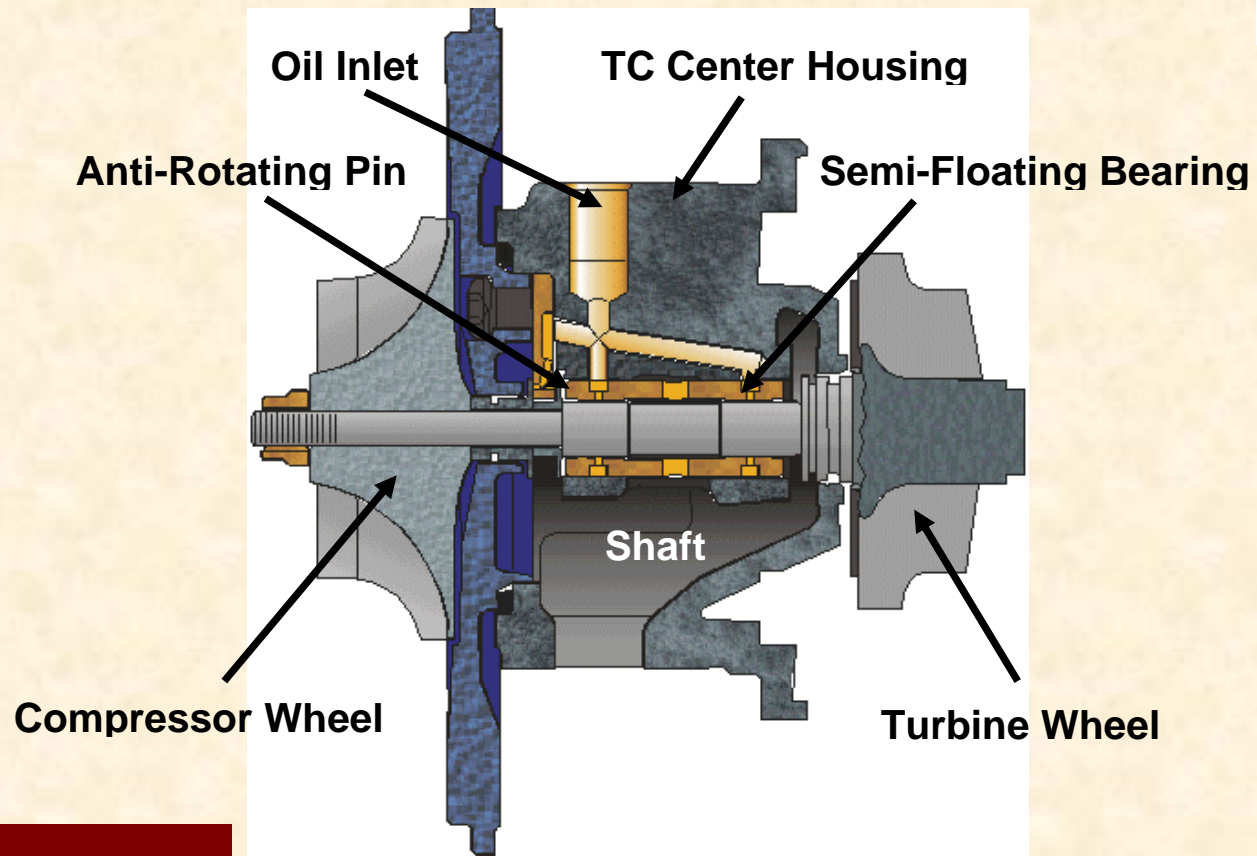
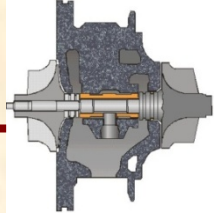


- Introduction to turbocharger rotordynamics
- **Experimental facilities**
- **Development of predictive models (Virtual Tool)**
- **Comparisons predictions vs test data**
- **Closure**



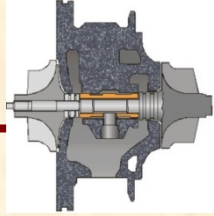


TC shaft motions virtual tool

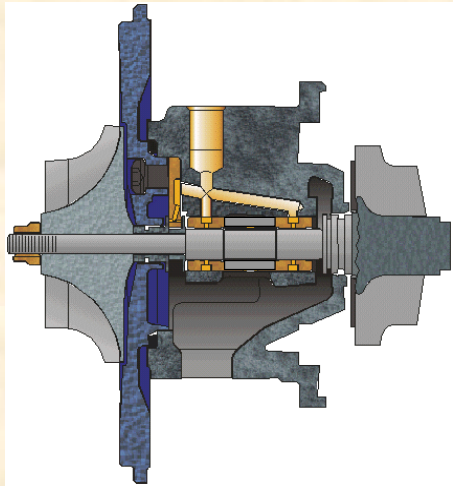


Turbochargers

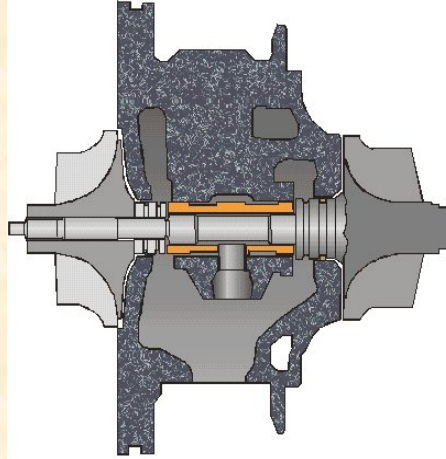
- Increase internal combustion (IC) engine **power** output by forcing more air into cylinder
- Aid in producing **smaller**, more **fuel-efficient** engines with **larger power** outputs



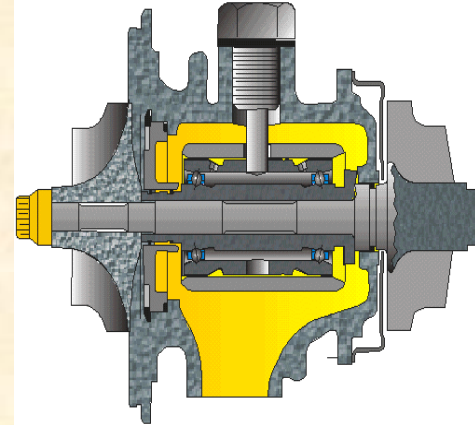
RBS: TC Rotor Bearing System(s)



RBS
Fully Floating Bearing



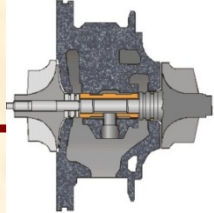
RBS
Semi Floating Bearing



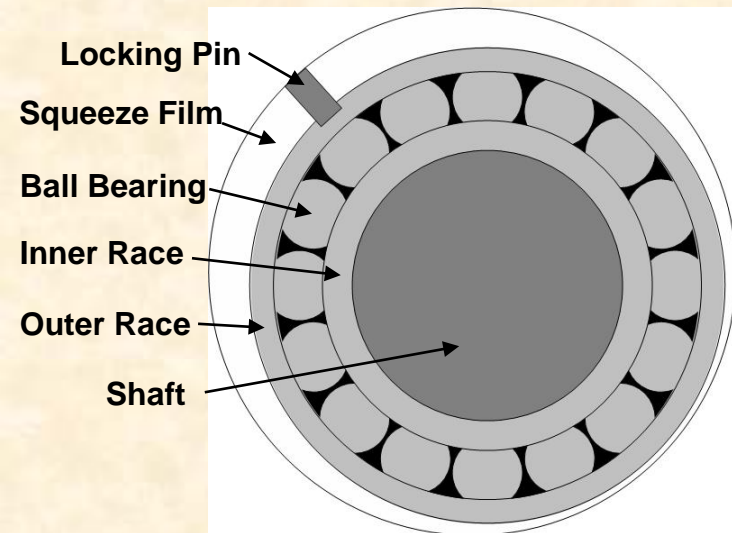
RBS
Ball Bearing

The driver:

Increased IC engine performance & efficiency demands of robust & turbocharging solutions

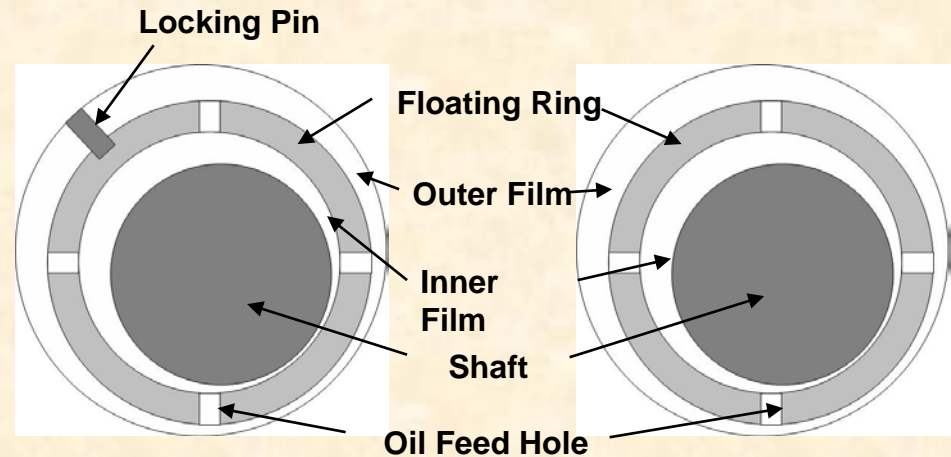


Bearing types



Ball-Bearing

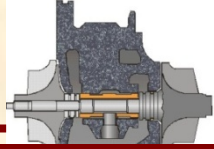
- **Low shaft motion**
- **Relatively expensive**
- **Limited lifespan**



Semi-Floating Ring Bearing (SFRB)

Floating Ring Bearing (FRB)

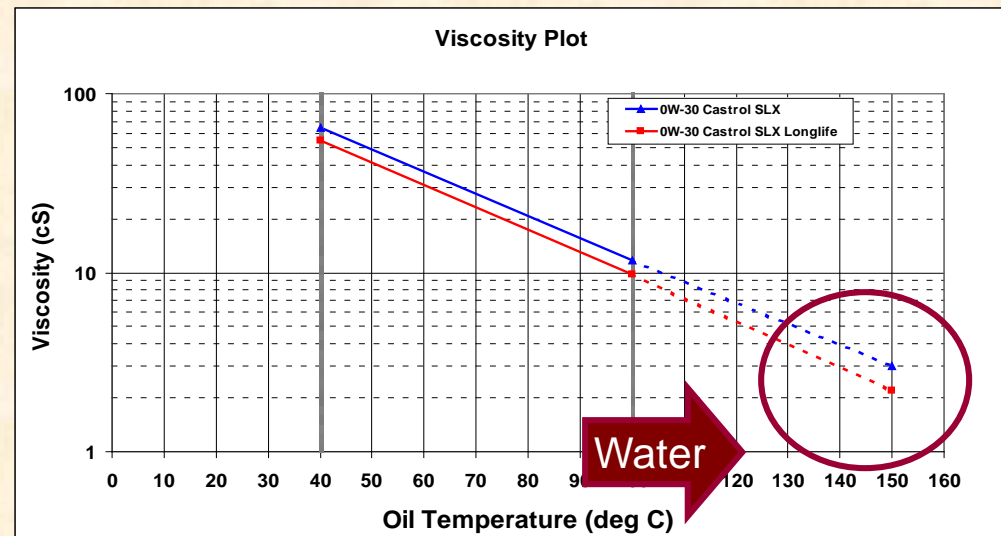
- **Economic**
- **Longer life span**
- **Prone to subsynchronous whirl**



Major challenges: extreme operating conditions

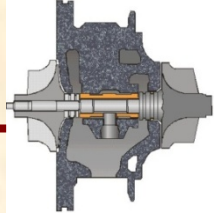
- - Low Oil Viscosity, e.g. 0W30 or 0W20
- - High Oil Temperature (up to 150°C)
- - Low HTHS (2.9); Low Oil Pressure (1 bar)
- - Increased Maximum Turbocharger Speed
- - Variable Geometry Turbo Technology & Assisted e-power start up
- - High Engine Vibration Level
- - More Stringent Noise Requirements

Need predictive tool to reduce costly engine test stand qualification





Literature Review: San Andres and students



- TC linear and nonlinear rotordynamic codes – GUI based – including engine induced excitations
- **Realistic bearing models: thermohydrodynamic**
- Novel methods to estimate imbalance distribution and shaft temperatures
- **NL analysis for frequency jumps and noise reduction**
- Measured ring speeds with fiber optic sensors

2004	IMEchE J. Eng. Tribology
2005	ASME J. Vibrations and Acoustics ASME DETC 2003/VIB-48418 ASME DETC 2003/VIB-48419
2007	ASME J. Eng. Gas Turbines Power ASME GT 2006-90873
2007	ASME J. Eng. Gas Turbines Power ASME GT 2005-68177
2007	ASME J. Tribology IJTC 2006-12001
2007	ASME DETC2007-34136
2010	ASME J. Eng. Gas Turbines Power ASME GT2009-59108
2010	IFToMM Korea

TC testing: expensive and time consuming

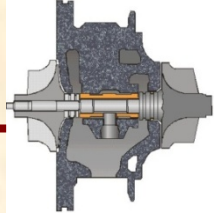


Predictive tool saves time and money

Benchmarked against test data

Predictive tool for shaft motion **benchmarked by test data**

Main Tasks – KEY OBJECTIVES



1. Measure shaft motion response in dedicated PV and CV turbocharger test rigs (cold & hot gas)

Test rigs

2. Development of software for prediction of (S) floating ring bearing static and dynamic forced response

XLBRG

3. Integration of FRB and SFRB tools into nonlinear rotordynamics code – VIRTUAL LABORATORY

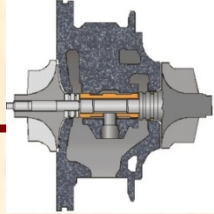
XLTRC₂

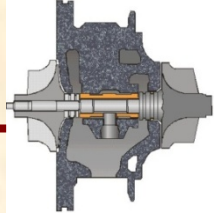
4. Comparisons of test data to predictions: Validate predictive tool

Tools

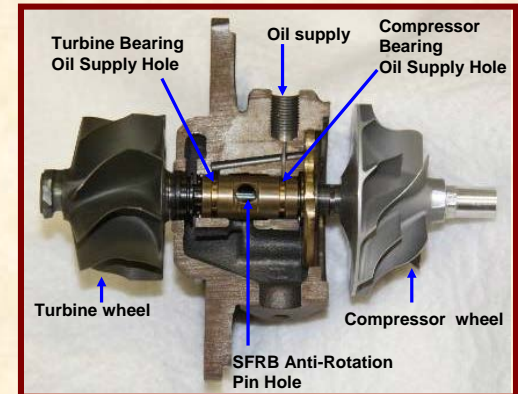


TC shaft motions virtual tool





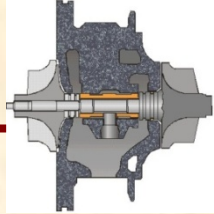
Test rigs for TC rotordynamic performance evaluation



KEY OBJECTIVE # 1



TC shaft motions virtual tool



Experiments to measure the rotordynamic response of a turbocharger supported on semi-floating ring bearings and fully floating ring bearings

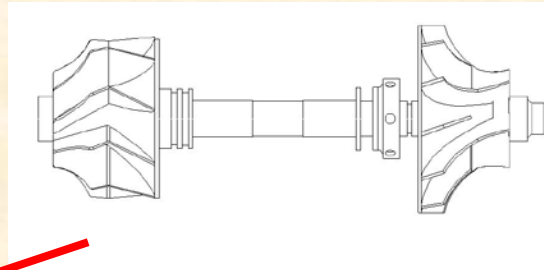
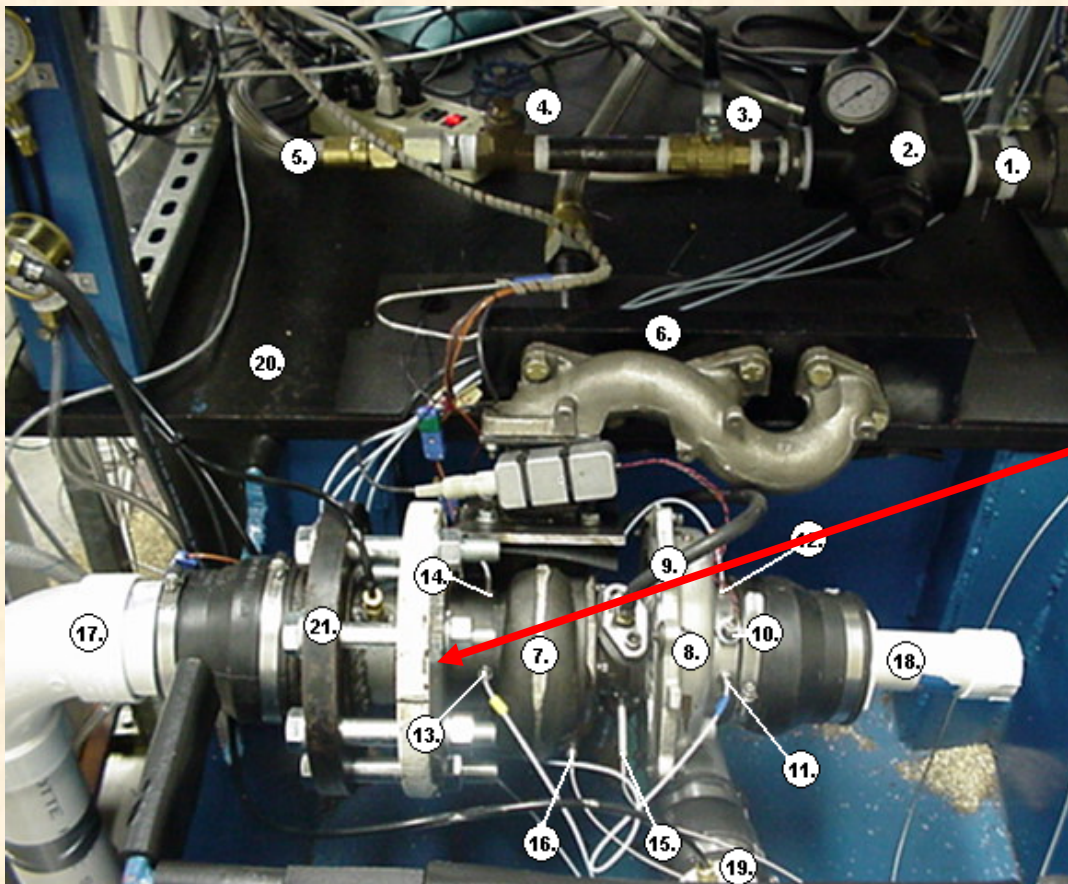
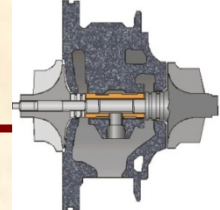
Test Rigs

Construct various test rigs, develop measurement methods, strategy to sensor selection and measurement locations, acquire data, processing tools, etc

KEY OBJECTIVE # 1



TC shaft motions virtual tool

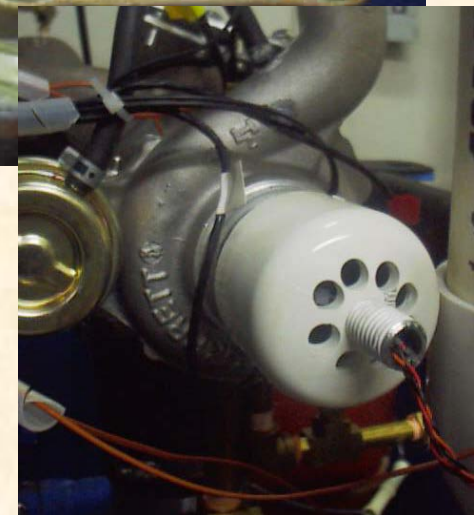
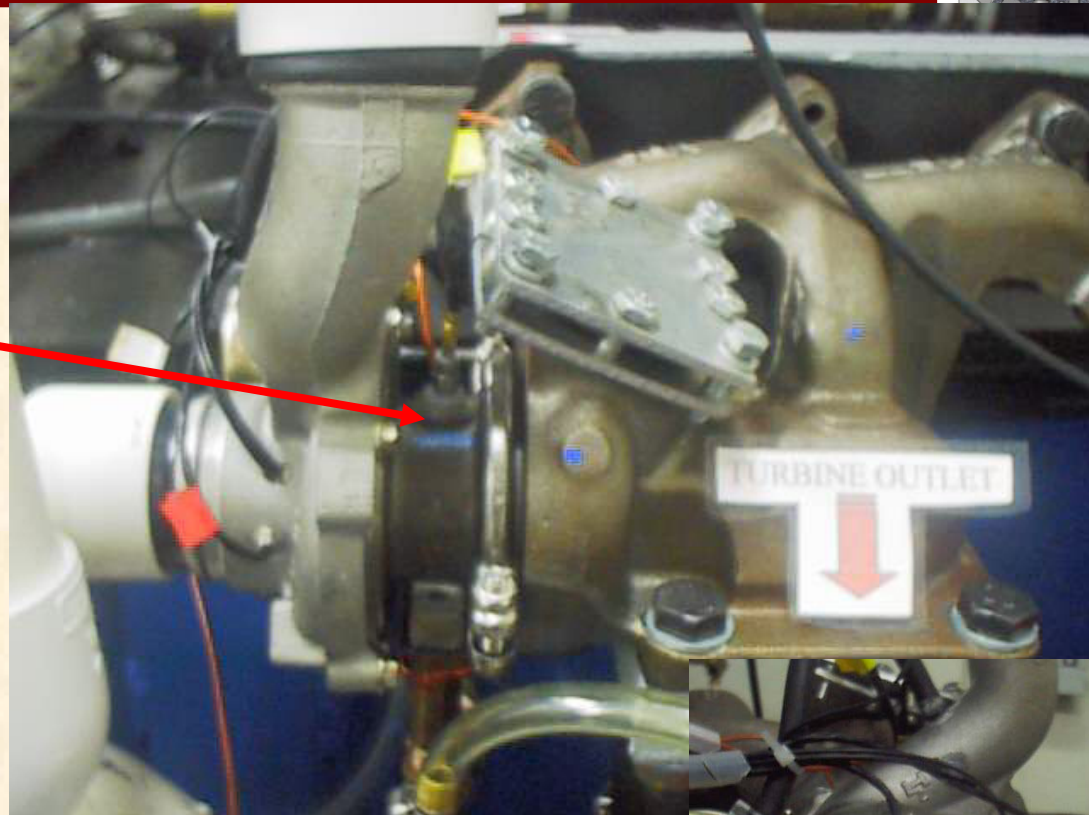
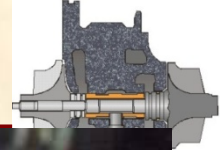


- | | | |
|-------------------------------|---|---|
| 1. High pressure supply | 10. Infrared tachometer | 15. Fiber optic floating ring tachometer (Compressor End) |
| 2. Pressure regulator | 11. Compressor X-Direction Proximity Transducer | 16. Fiber optic floating ring tachometer (Turbine End) |
| 3. Shut-off valve | 12. Compressor Y-Direction Proximity Transducer | 17. Turbine outlet |
| 4. Turbine flow control valve | 13. Turbine X-Direction Proximity Transducer | 18. Compressor inlet |
| 5. Flexible Hose | 14. Turbine Y-Direction Proximity Transducer | 19. Compressor Outlet |
| 6. Supply Manifold | | 20. Steel Table |
| 7. Turbine | | 21. Turbine Outlet Safety Structure |
| 8. Compressor | | |
| 9. Lubricant Supply | | |

- Infrared tachometer
- RAM BN sensors for shaft motion
- Fiber optics for ring motion detection



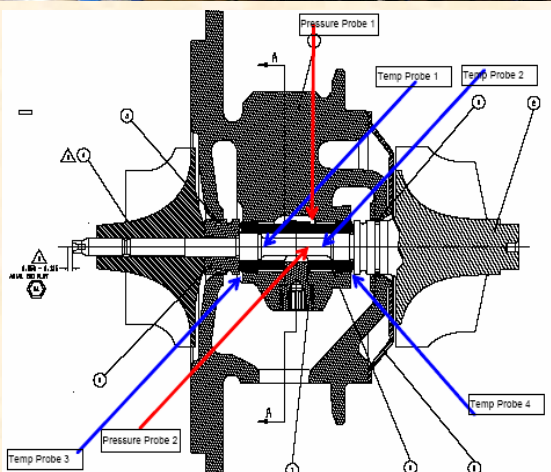
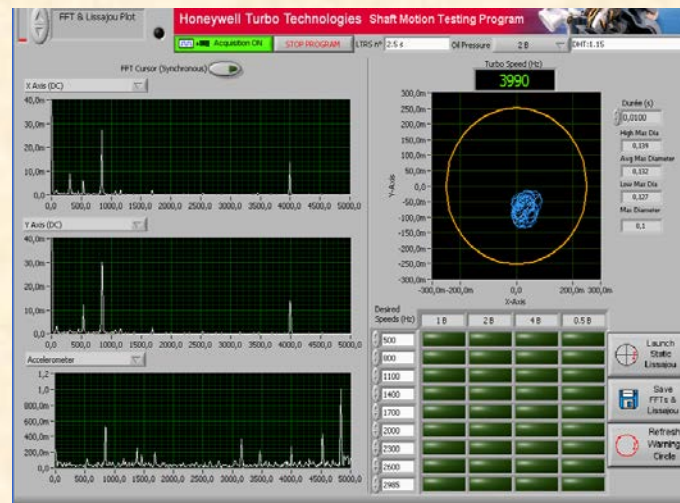
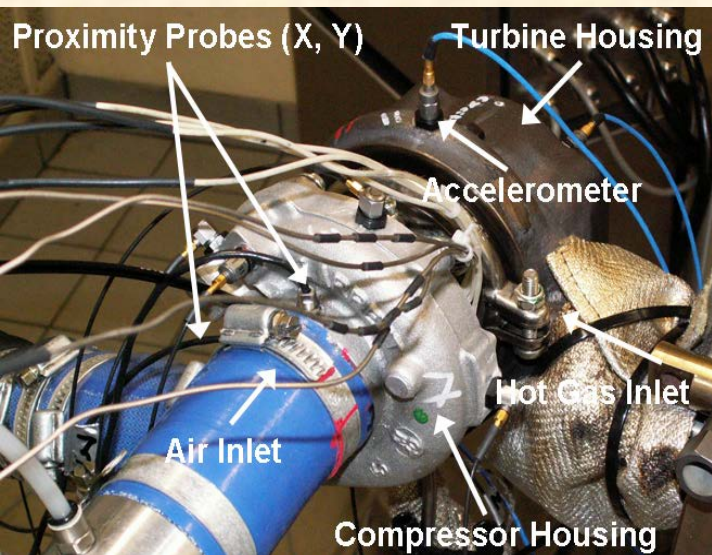
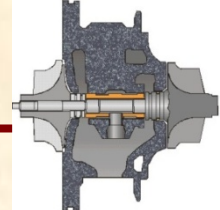
TC shaft motions virtual tool



- Infrared tachometer
- KAMAN sensors for shaft displacement at compressor side
- Accelerometers for casing motion
- 240 krpm max (4 KHz)



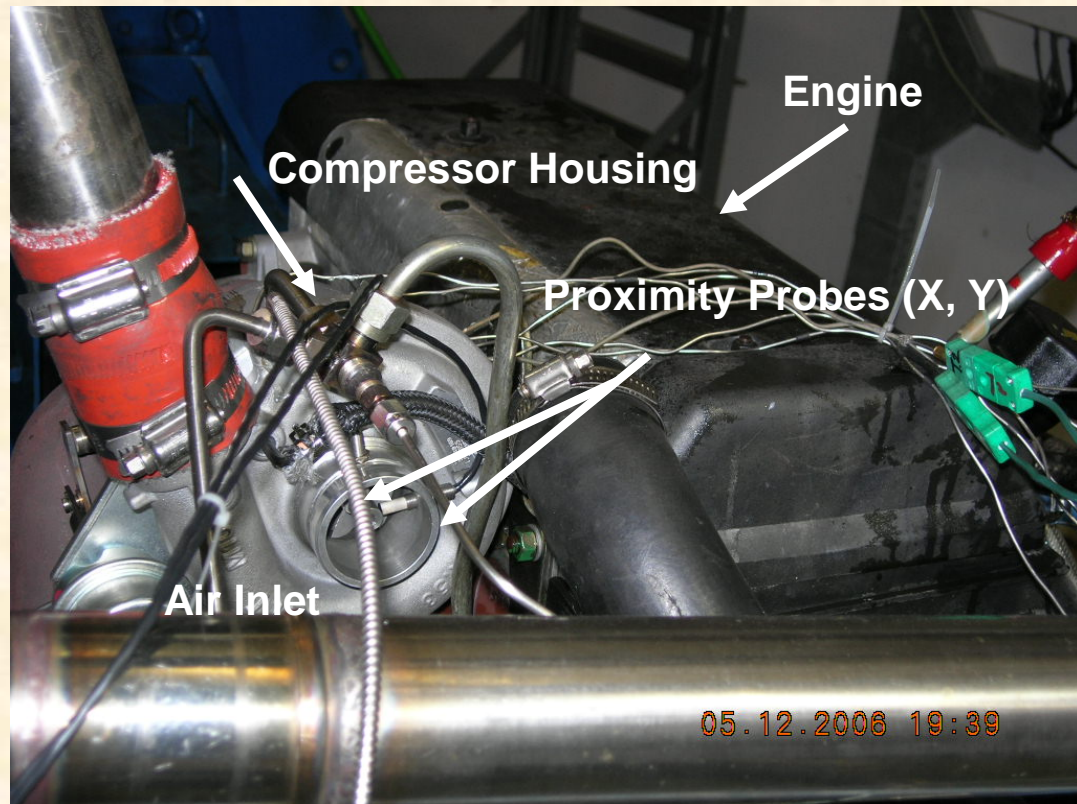
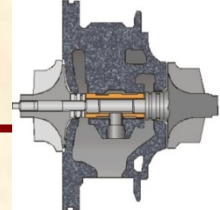
TC shaft motions virtual tool



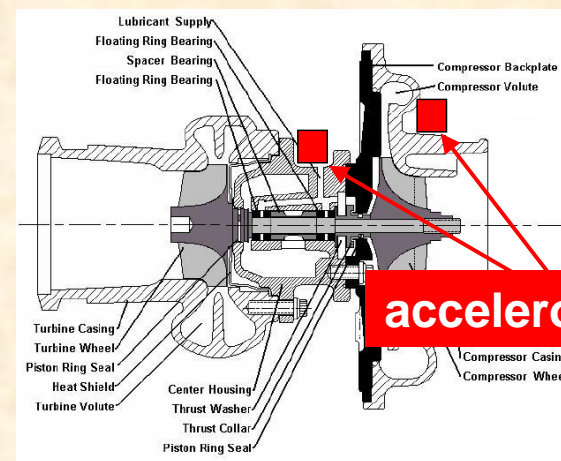
- KAMAN sensors for shaft displacement at compressor side
- connection to shakers
- 300 krpm max (5 KHz)



TC shaft motions virtual tool



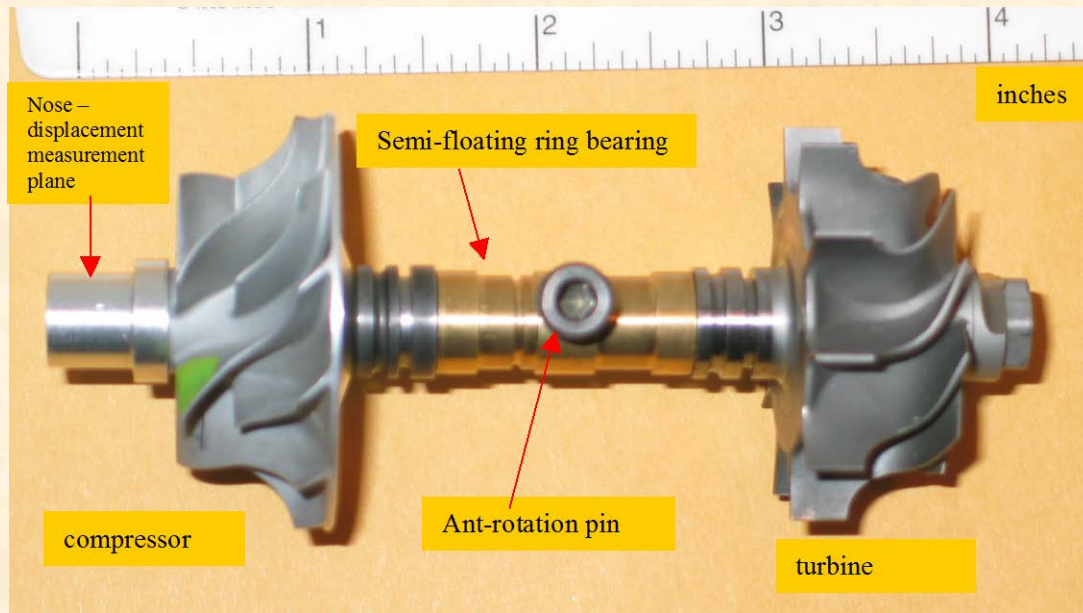
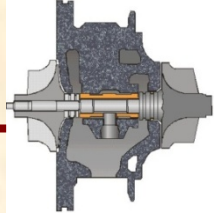
**3-axes
accelerometers:
engine isolated atop
a large shaker table**



TC engine stand test rig—HTT (Shanghai) 2008



TC shaft motions virtual tool

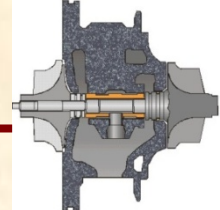


Shaft speed **25 - 240** krpm, Oil 5W-30,
150 C inlet temperature, feed pressure 1- 4 bar

Measure rotordynamic response of PV turbocharger

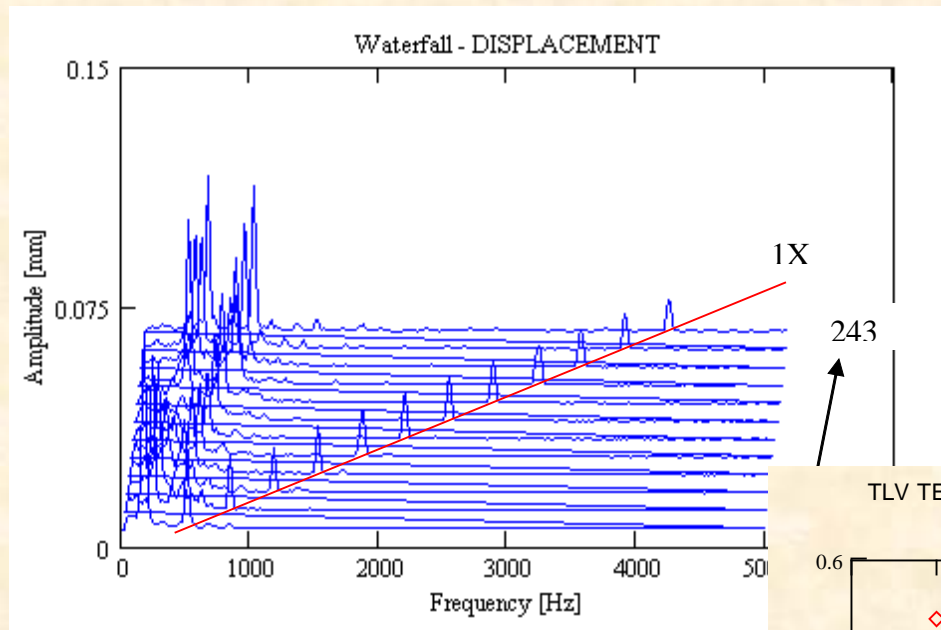


TC shaft motions virtual tool



TEST DATA - Compressor End

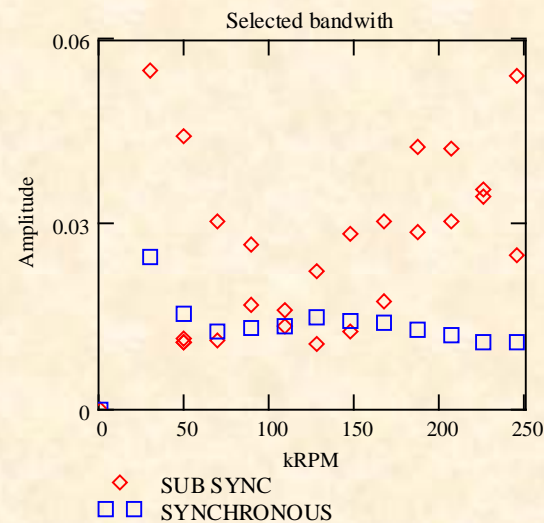
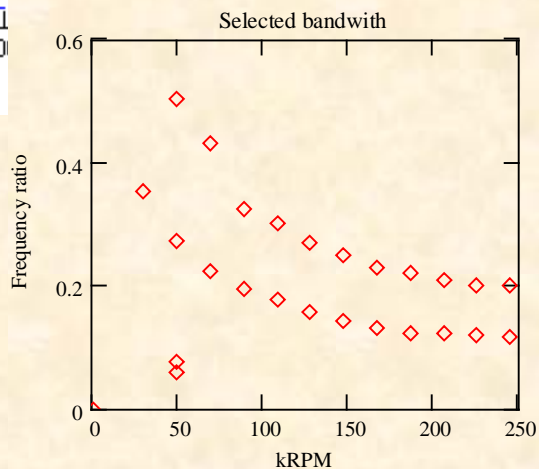
TLV TEST DATA



Dominance of sub synchronous motions at all speeds

TLV TEST DATA - compressor end

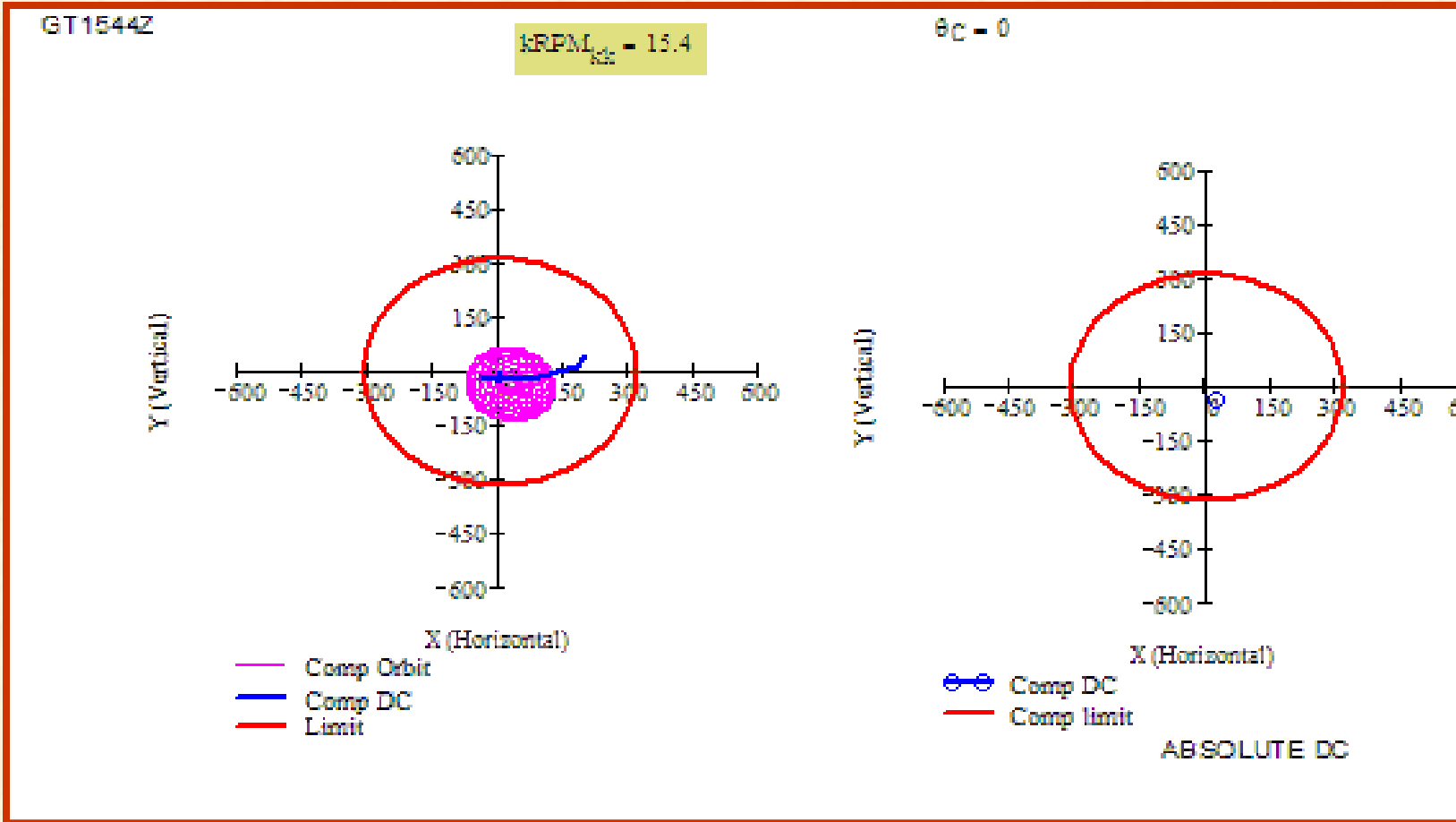
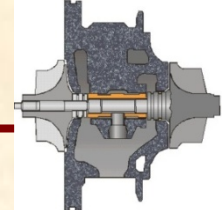
Amplitude (mm)



waterfall compressor end shaft motions
whirl frequency ratio and amplitudes (mm) of vibration. Oil supply pressure = 1 bar, T=150 C



TC shaft motions virtual tool

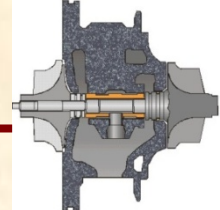


TAMU TEST DATA

TC failure (cold air operation) 10 - 110 krpm : Oil ISO VG 10

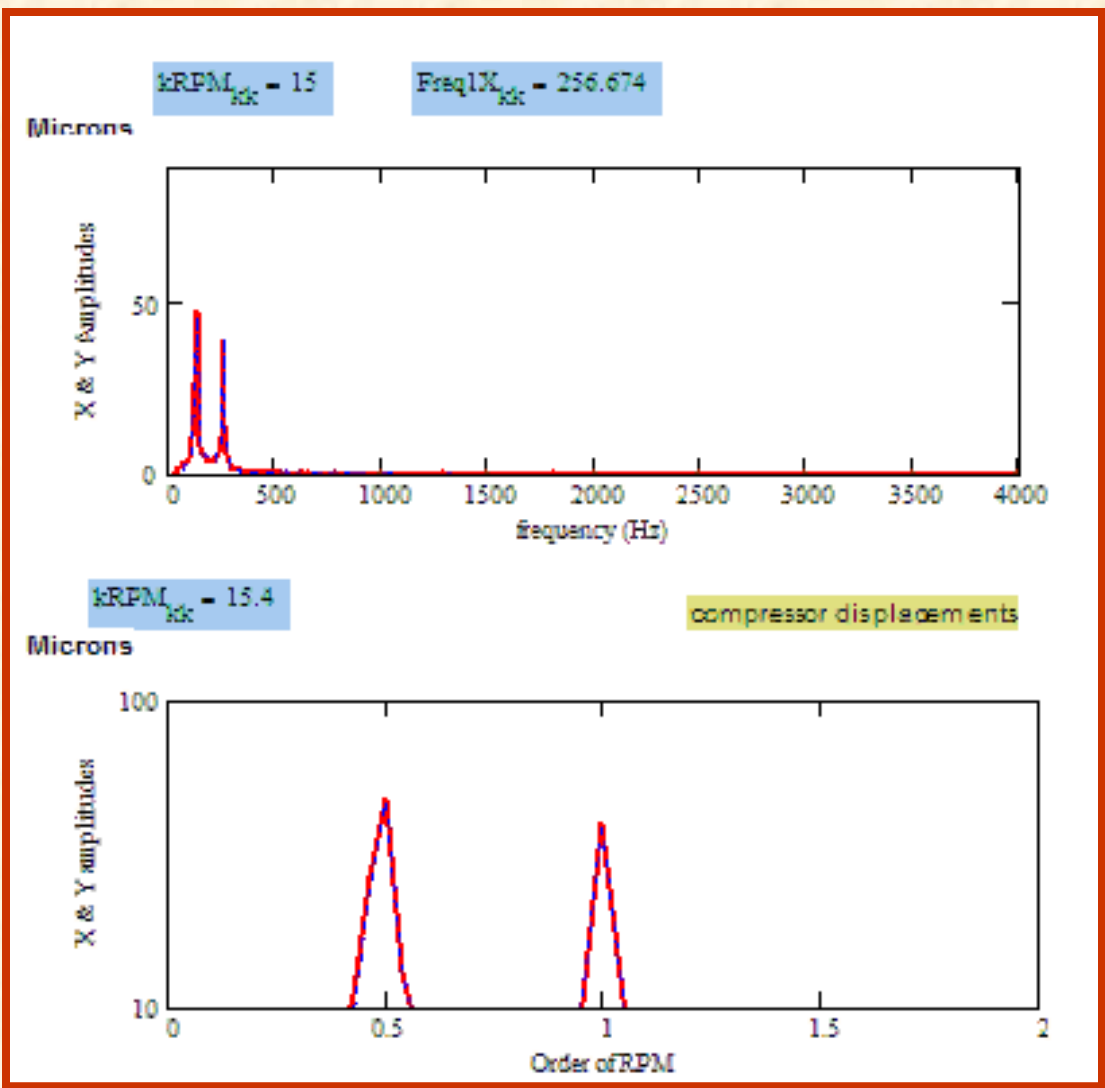


TC shaft motions virtual tool



Purpose of analysis is to reduce risk for this type of failure

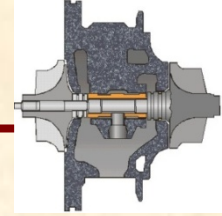
TAMU TEST DATA



TC failure (cold air operation) 10 - 110 krpm : Oil ISO VG 10

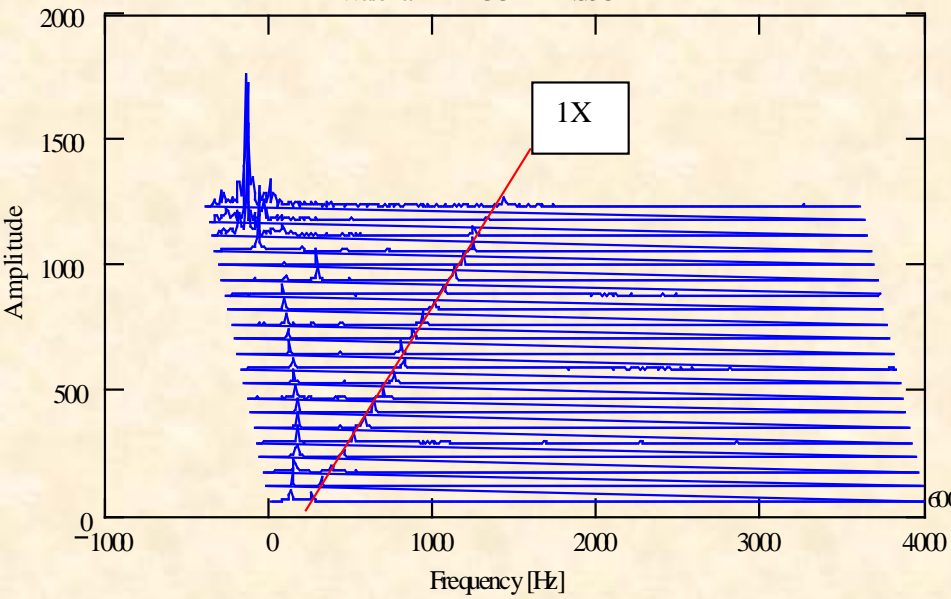


TC shaft motions virtual tool



TAMU TEST DATA

Waterfall - X-COMPRESSOR



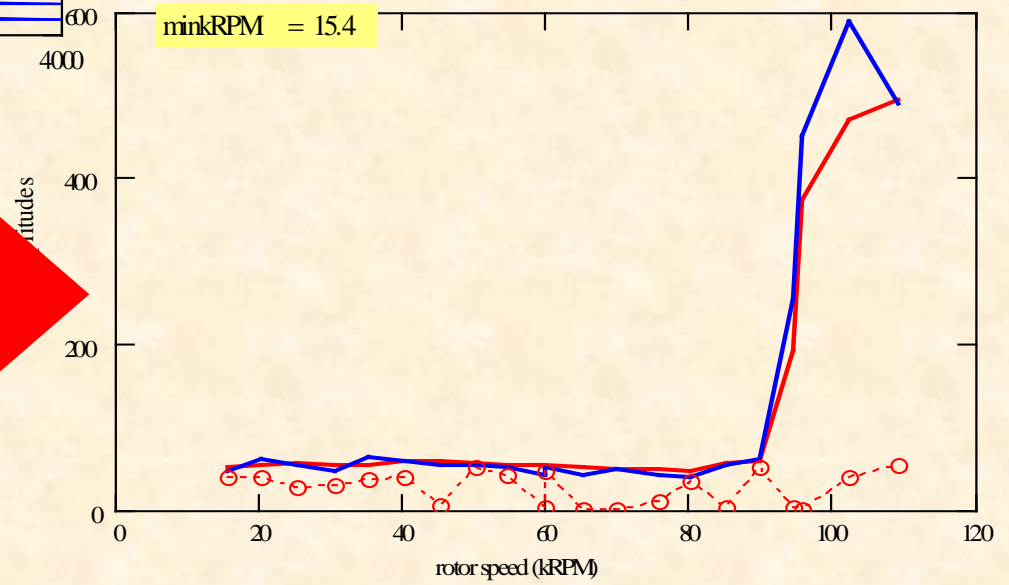
$CX_{max} = 589.28$

microns

maxkRPM = 109

TC failure

C-X motions



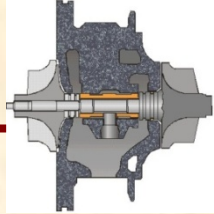
minRPM = 15.4

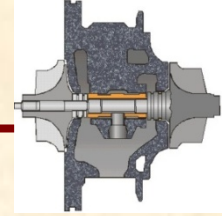
- Comp Overall
- Comp Peak
- Comp at WFR

(cold air operation)
10 - 110 krpm : Oil ISO VG 10

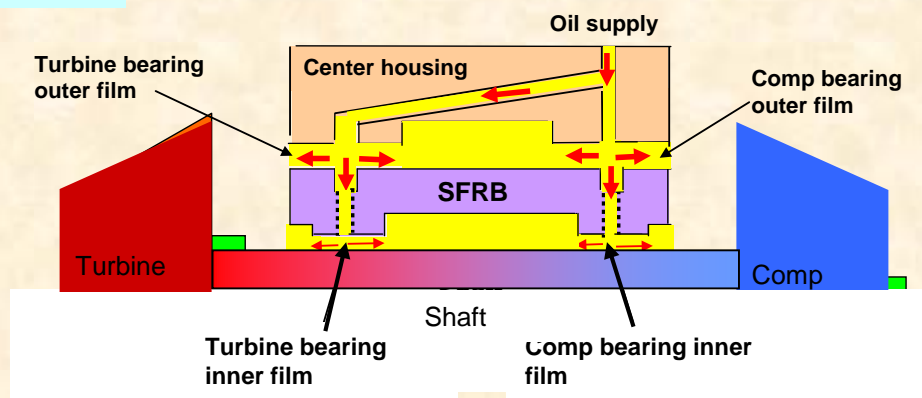
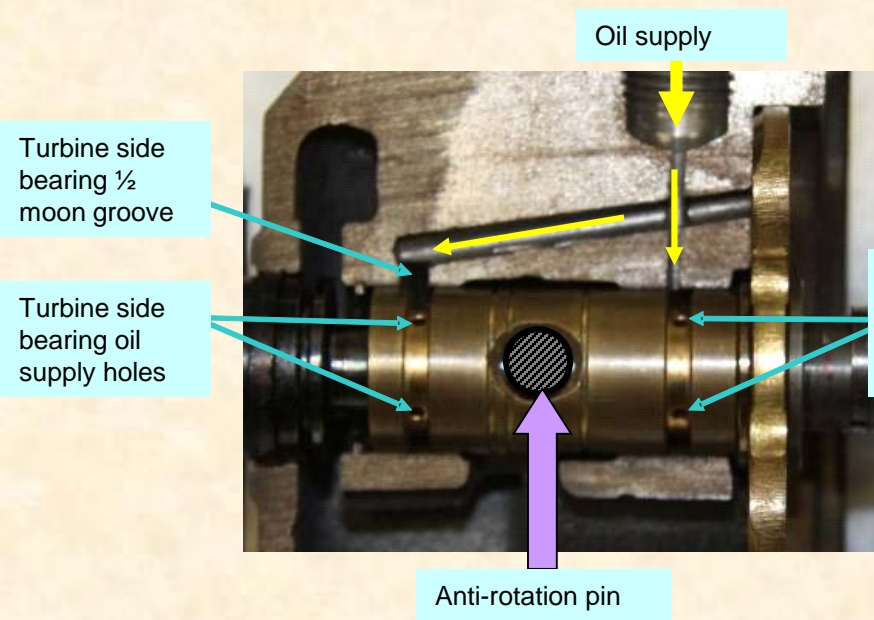


TC shaft motions virtual tool



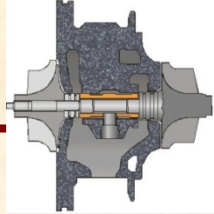


TC fluid film bearings





TC shaft motions virtual tool



Development of software for prediction of (semi) floating ring bearing (S-FRB) static and dynamic forced response

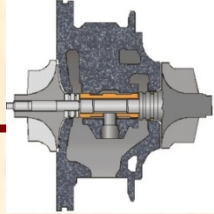
XLBRG Tool

EXCEL & Fortran FEM code for prediction of FRBs and SFRBs forced response (static and dynamic)

Finite length bearing model with global thermal balance and shear thinning effects

Interface to XLTRC² software for rotordynamics analysis

KEY OBJECTIVE # 2



Reynolds Equations

Outer film pressure, P_o

$$\nabla \left\{ \frac{h_o}{12\mu_o} \nabla (P_o) \right\} = \left\{ \dot{e}_{X_R} + e_{Y_R} \frac{\Omega_R}{2} \right\} \cos \Theta + \left\{ \dot{e}_{Y_R} - \frac{\Omega_R}{2} e_{X_R} \right\} \sin \Theta$$

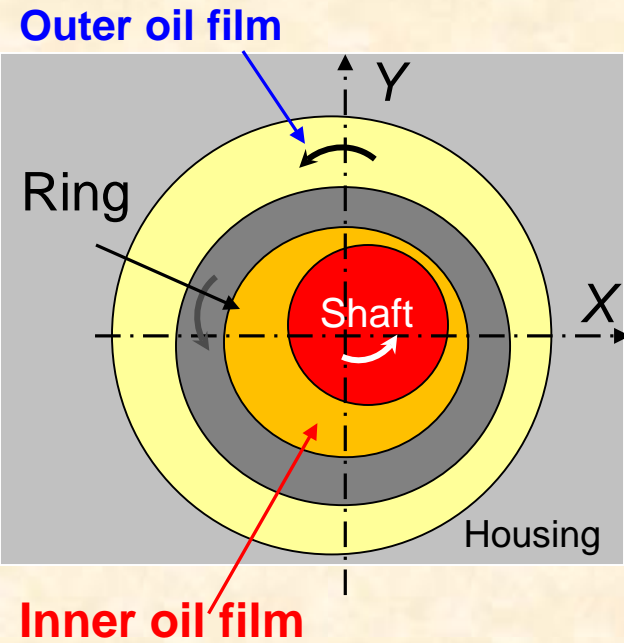
Film thickness: $h_o = c_o + e_{X_R} \cos(\Theta) + e_{Y_R} \sin(\Theta)$

Inner film pressure, P_i

$$\nabla \left\{ \frac{h_i}{12\mu_i} \nabla (P_i) \right\} = \left\{ \delta \dot{e}_X + \delta e_Y \frac{\Omega_+}{2} \right\} \cos \Theta + \left\{ \delta \dot{e}_Y - \frac{\Omega_+}{2} e_X \right\} \sin \Theta$$

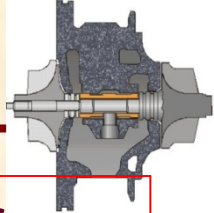
Film thickness: $h_i = c_i + \delta e_X \cos(\Theta) + \delta e_Y \sin(\Theta)$

$$\delta e_X = e_{X_J} - e_{X_R}; \delta e_Y = e_{Y_J} - e_{Y_R}; \Omega_+ = \frac{\Omega_J + \Omega_R}{2}$$



- **Balance of drag torques** from outer and inner oil films
- **Thermal energy transport** (heat conduction & convection)

Lumped Parameter Thermal Model



Oil energy increase

~ Heat flow

Sp Heat x Mass flow x
Temperature Difference

bearing

Mechanical power

by fluid shearing

$P \sim \text{Torque} \times \text{Rot Speed}$

Outer film Temp Rise

Inner film Temp Rise

Outer film

Inner film

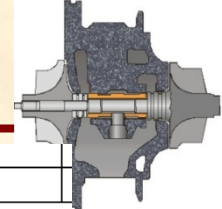
Energy convected
to solids and
conducted through
shaft, ring and
bearing

Floating
ring

shaft



XLBRG® INPUT



Select

Thermal Model	0
Heat convection model	1

Select

Geometry **INNER FILM**

REGULAR

Select

Geometry **OUTER FILM**

REGULAR

mn-mn	
mx-mn	

Select

Bearing Model 2

SEMI FLOATING RING

BUTTON PIN

LUBRICANT

Lub param	9	SAE 5W-30
Viscosity	0.00494	Pa-s
Density (kg/m³)	850	
Specific heat (J/kg-C)	1880	

Diameter (m)	0.0079
Lengths (m)	0.0046
Cold Radial Clearance (m)	7.50E-06
Hot Radial Clearance	9.76E-06

Diameter (m)	0.014
Lengths (m)	0.0062
Cold Radial Clearance (m)	3.50E-05
Hot Radial Clearance	2.72E-05

0.0000	Axial length [m]
0	Arc length [deg]
0	Circumferential l
0.0	WARNING: Value no

Pressures (bar)

Discharge	0	0
Cavitation	0	bar

Number pads	1	
Pad Arc Angle	360	deg
Leading Edge Pad 1	0	deg
Pad preload	0	-
Pad offset	0	

Select Bearing case

Fixed RING

0.85	Desired Static dis
3.71E+05	Recommend Stiff
9.11E+05	N/m
9.11E+05	N/m

Temperatures **COLD**

Turbine	700	°C
Defect temp ratio	0.13	

Operating

SHAFT	20.0	°C
195.4	°C	
CASING	150.0	°C
RING	172.7	°C
Inner-Outer		
Difference Inlet	0.0	°C

RING

Outer side groove depth	0.0000	m
Inner side groove depth	0.0000	m

Mass	0.0098	kg
Moment of inertia	3.10E-07	kg-m²
Weight	0.0581	N

Centrifugal pressure loss coefficients

Inner film	0.00
Outer film	0.00

Oil

Supply Press	4	N/bar	
Inlet Pressure Lo	0.9		
FEED Side load factors	5.43	5.73	
Load	5.43	5.73	
Compr side	5.43	5.73	
Turb Side	12.4	11.6	
Total	17.81	17.36	
high speed	240000	low speed	30000
typ speeds	240000	30000	5.779

MATERIAL PROPERTIES

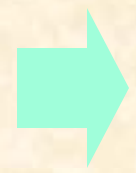
	Ring	Casing	Rotor
Thermal Coeff. Exp.	1.80E-05	1.25E-05	1.24E-05
Specific Heat	377.00	550.00	519.00
Conductivity	58.12	53.60	42.70
Density	8500.00	7150.00	7833.00
	brass	cast iron	steel

comment

Run XLBRG_v8.0 code

select

Update

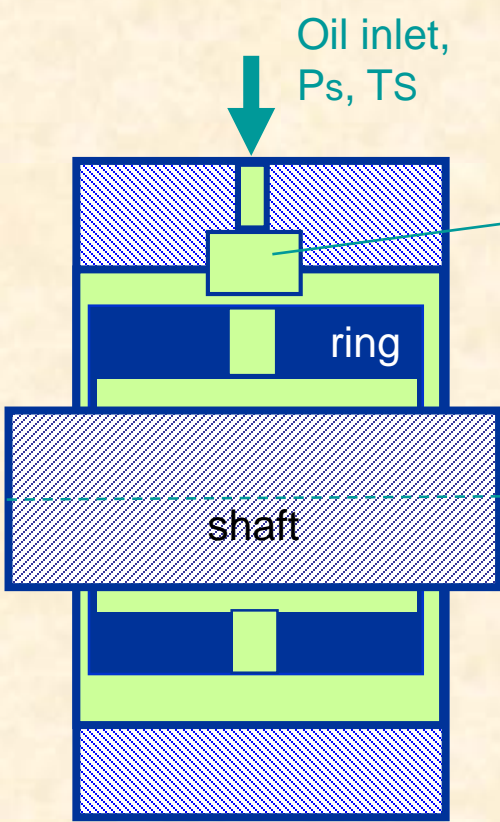
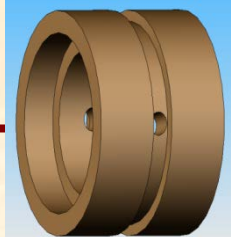


Geometry (cold) – L,D,C
 Fluid Type (commercial oil)
 Material properties
 Operation (speed and load)

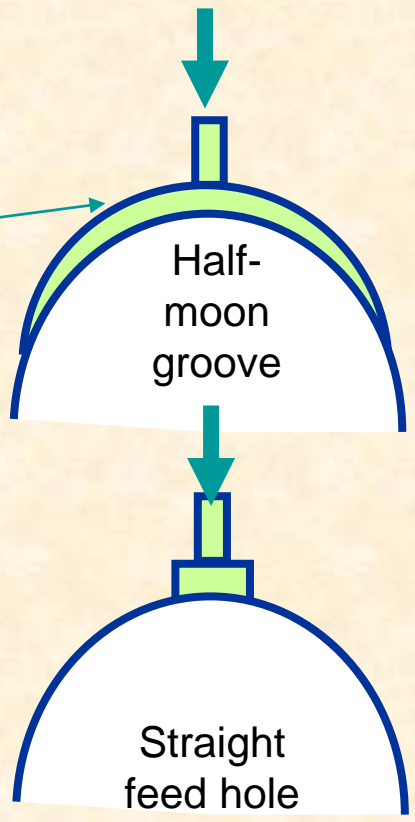
Example: Turbine side bearing



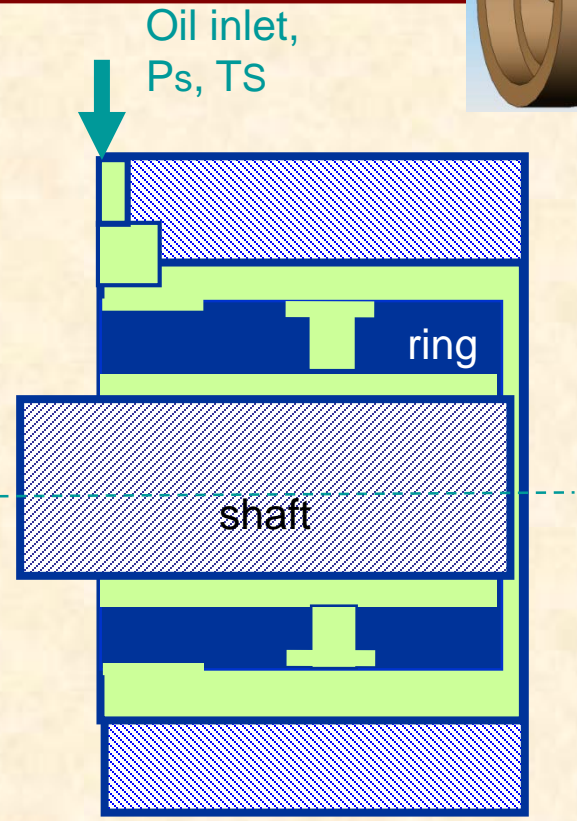
XLBRG®: types of bearings



Oil supply in bearing



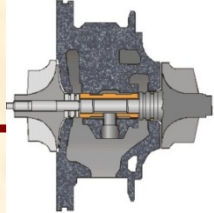
Types of oil supply



Oil supply – outboard side



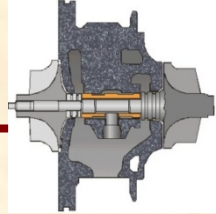
Figures NOT to scale



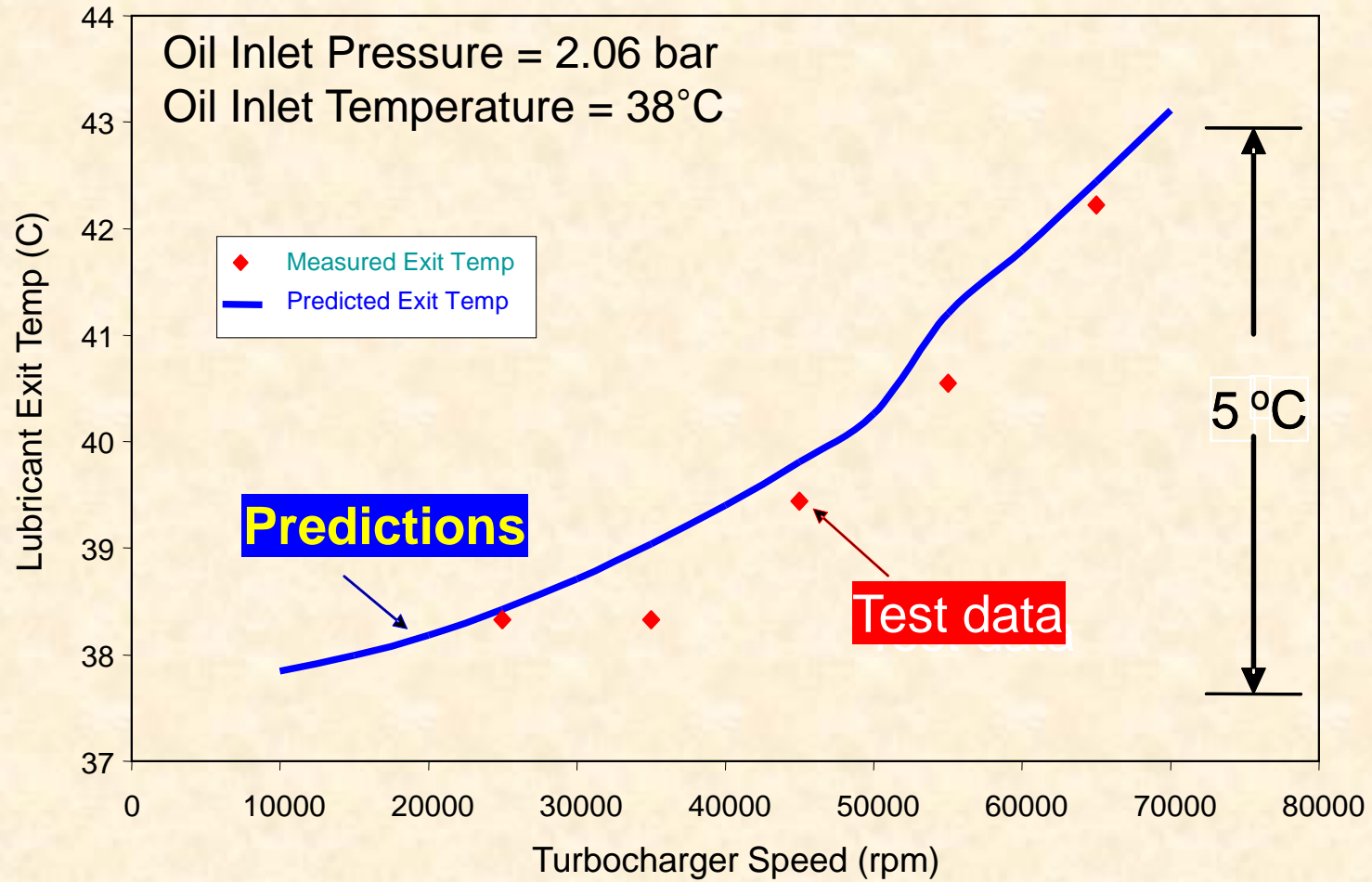
(Semi & Fully) Floating Bearing Ring

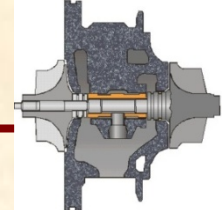
- **Actual geometry** (length, diameter, clearance) of inner and outer films, holes size and distribution
- **Supply conditions**: temperature & pressure
- **Lubricant viscosity** varies with temperature and shear rate (commercial oil)
- **Side hydrostatic load** due to feed pressure
- Temperature of casing
- Temperature of rotor at turbine & compressor sides derived from semi-empirical model: **temperature defect model**

XLBRG® ETHD fluid film bearing model predicts **operating clearance** and **oil viscosity** (inner and outer films) and **eccentricities** (static and dynamic) as a function of shaft & ring speeds and applied (static & dynamic) loads.



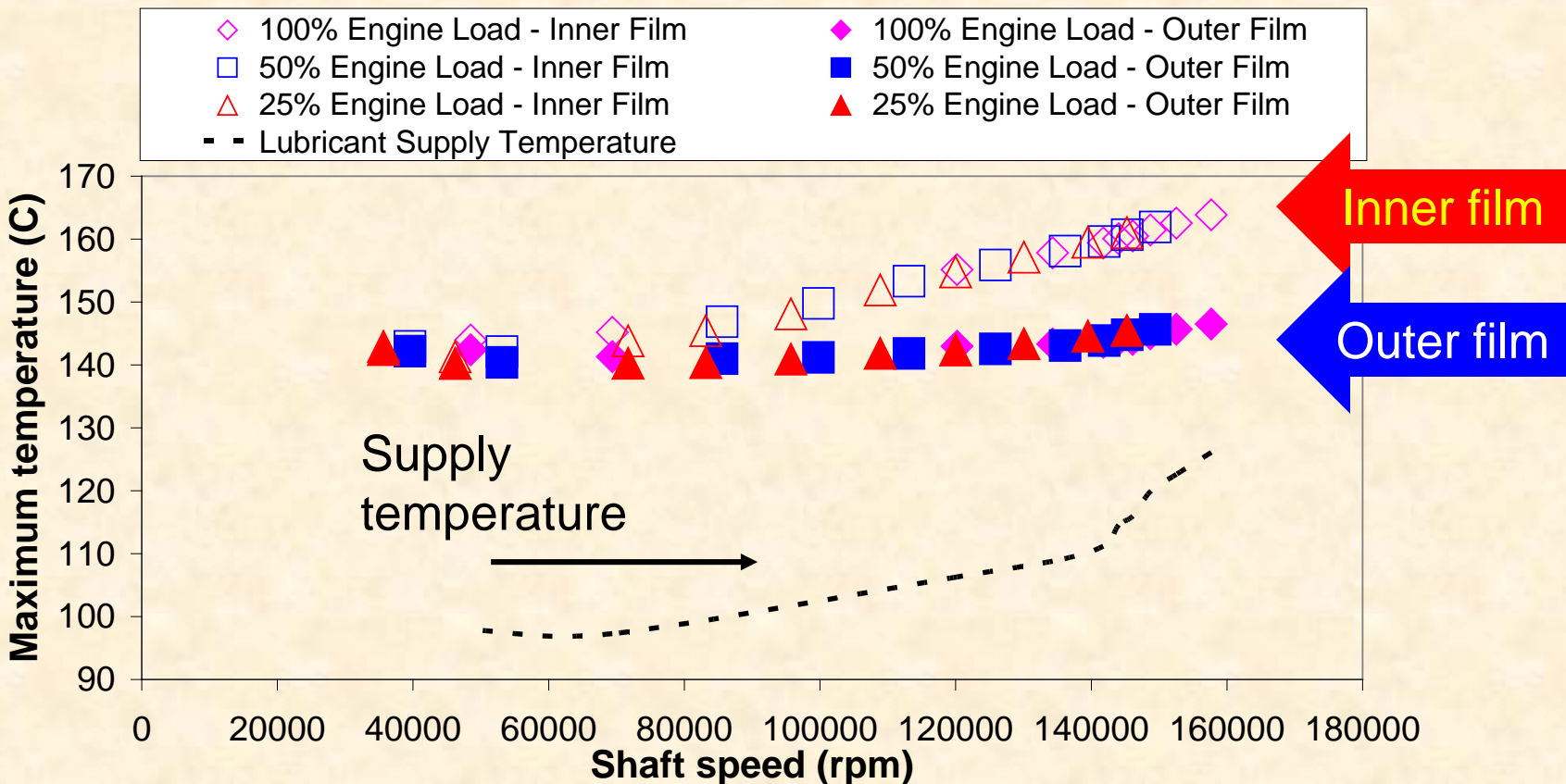
ASME GT2006-90873





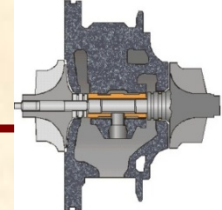
(S)FRB Predictions :

Peak film temperatures



Increase in power losses (with speed) leads to raise in inner film & ring temperatures.

No effect of engine load

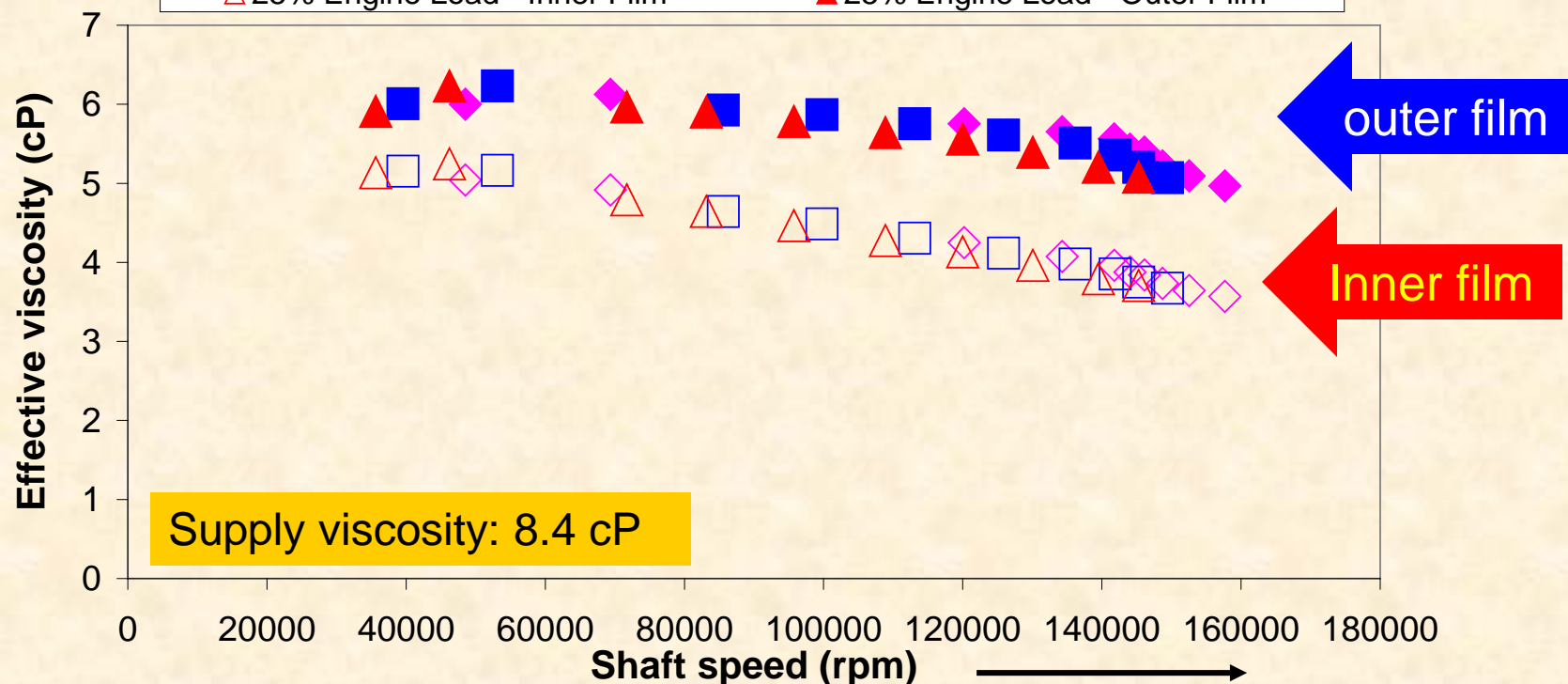


(S)FRB Predictions :

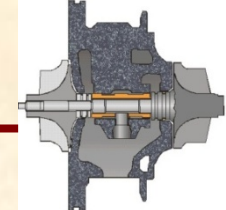
Oil effective viscosity

Lubricant type:
SAE 15W - 40

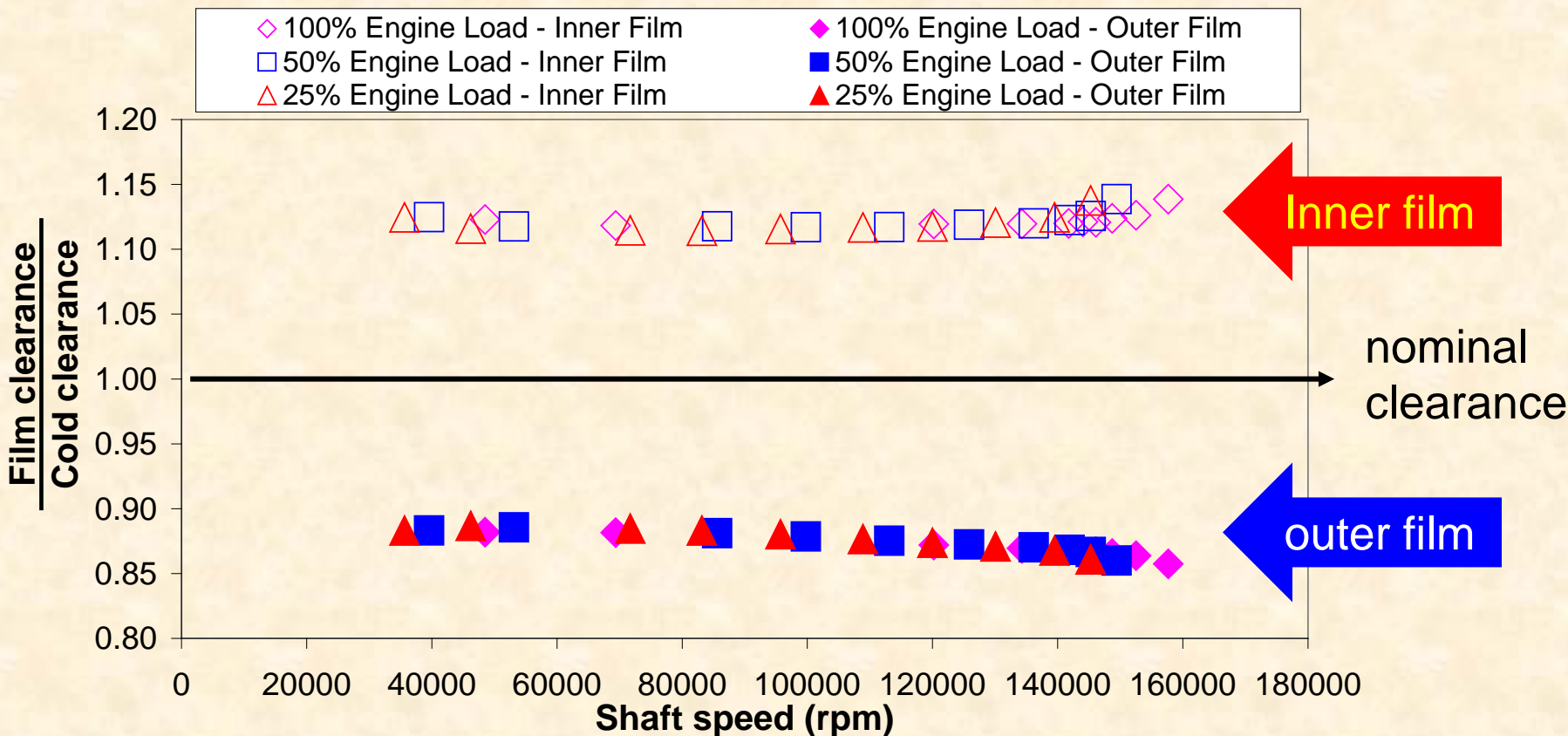
- ◇ 100% Engine Load - Inner Film
- ◇ 100% Engine Load - Outer Film
- 50% Engine Load - Inner Film
- 50% Engine Load - Outer Film
- △ 25% Engine Load - Inner Film
- △ 25% Engine Load - Outer Film



Higher film temperatures determine lower lubricant viscosities. Operation parameters independent of engine load



(S)FRB Predictions : Film clearances

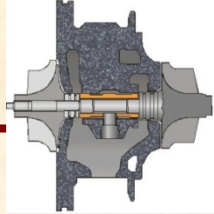


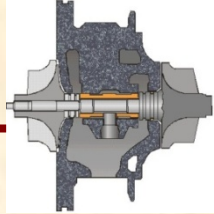
Thermal growth relative to nominal inner or outer cold radial clearance

Inner film clearance grows and outer film clearance decreases – RING grows more than SHAFT and less than CASING. Material parameters are important



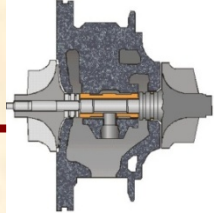
TC shaft motions virtual tool





TC rotordynamics

linear and nonlinear



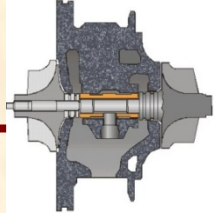
Integration of FRB and SFRB codes into nonlinear rotordynamics program

XLTRC² Rotordynamics Virtual Tool

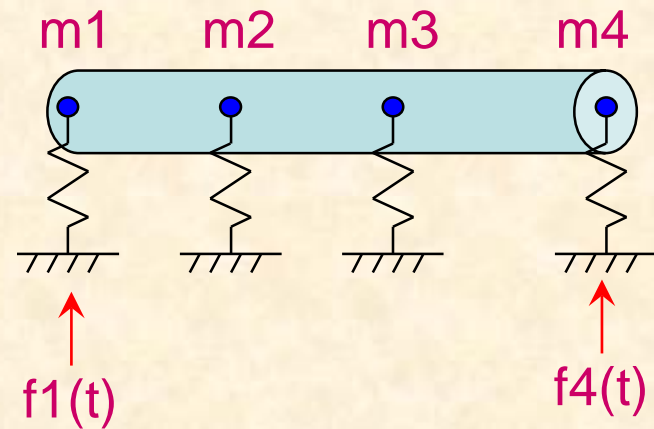
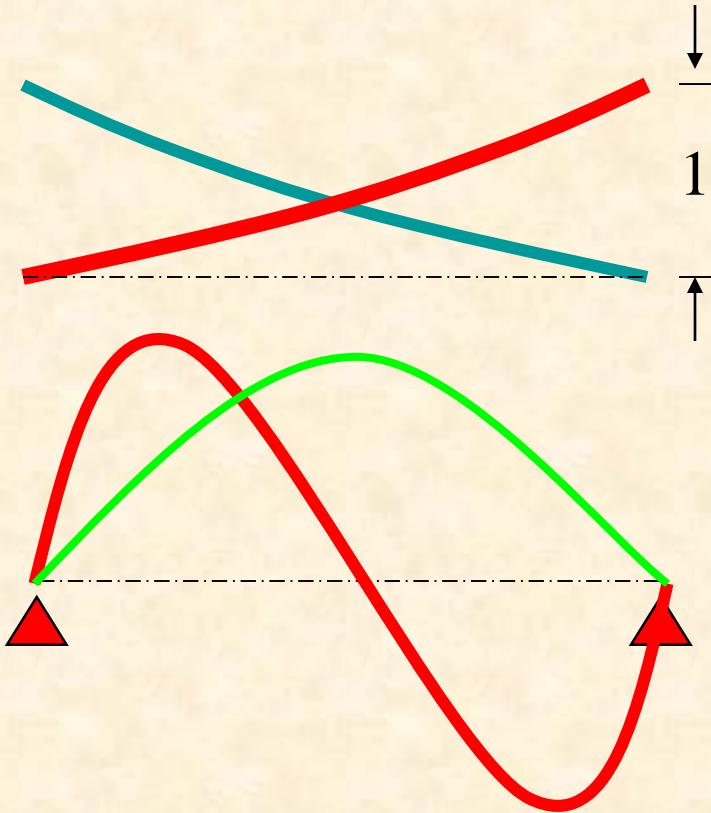
- Beam Finite-Element Formulation
- Real Component-Mode Synthesis (CMS) model
- Multi-line Rotor/Housing Modeling Capability
- **Linear and transient response nonlinear analyses**
- Fully integrated with an extensive suite of support codes
- User-Friendly GUIs for rapid model development and report generation

General EOMs

$$\mathbf{M} \ddot{\mathbf{q}} + \mathbf{C} \dot{\mathbf{q}} - \Omega \mathbf{G} \dot{\mathbf{q}} + \mathbf{K} \mathbf{q} = \mathbf{Q}(t)$$

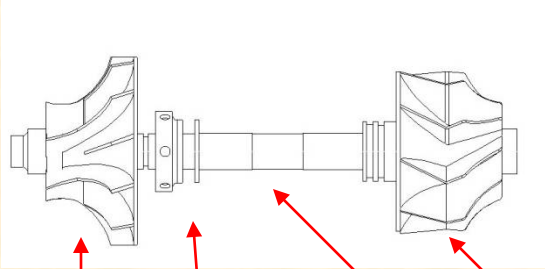
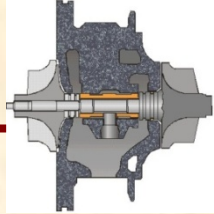


- Timoshenko-beam, FE-formulation
- Calculates real modes
- Reduces model dimensionality by using a limited number of modes



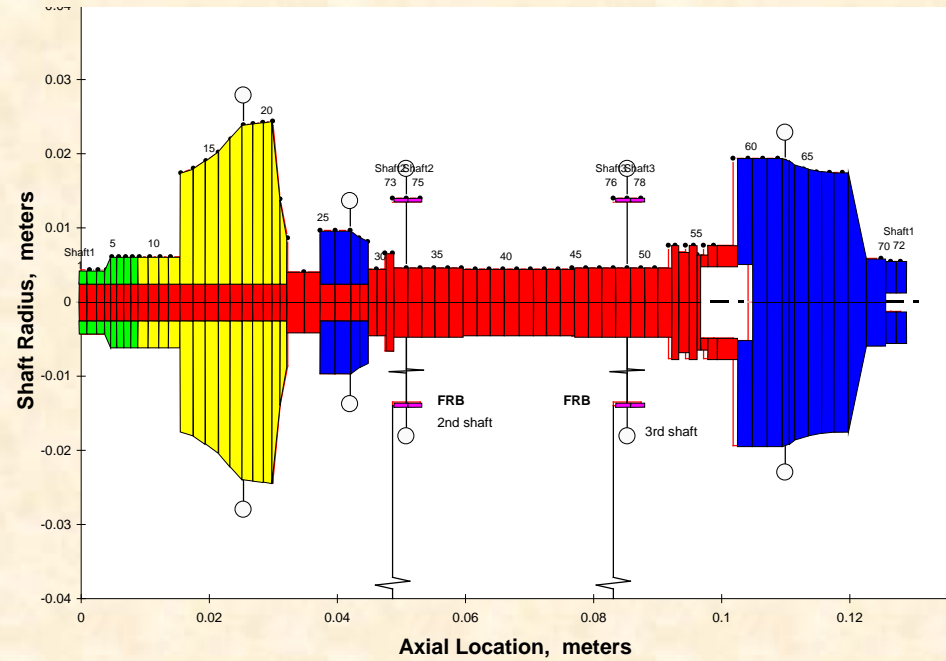


Rotor structural FE models



Compressor thrust disk shaft turbine

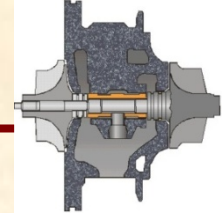
Typical TC rotor hardware



Typical FE rotor structure model

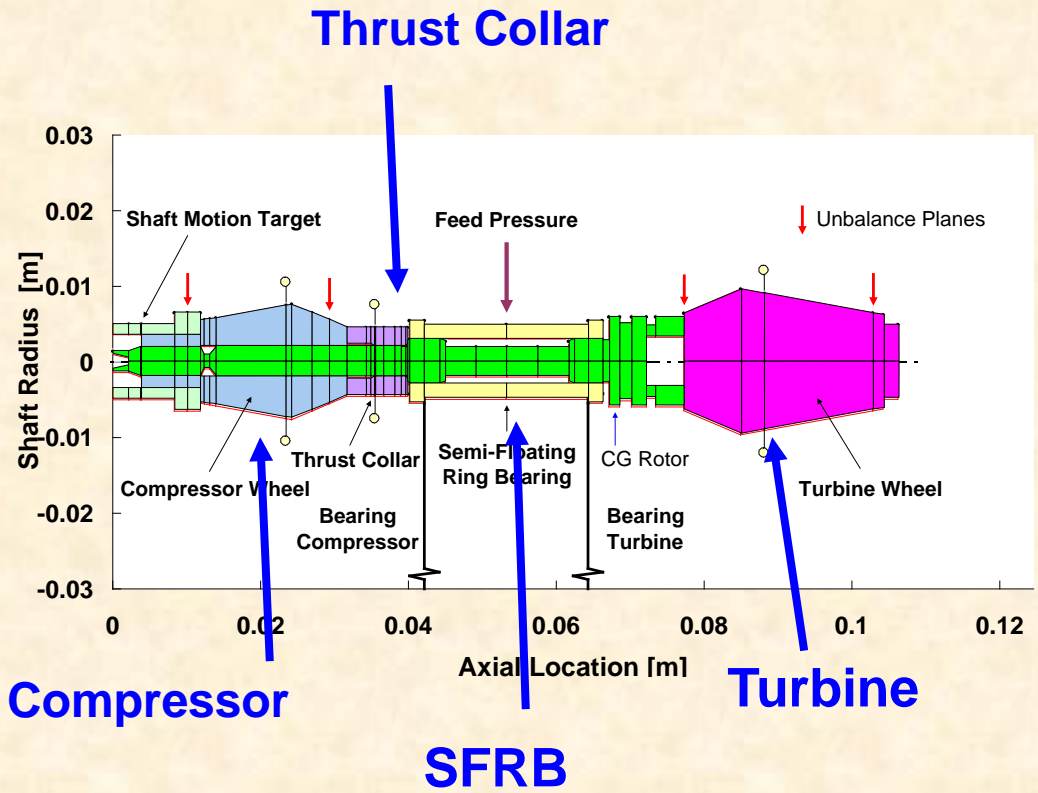


Validate rotor model



Rotor finite element model: 2 shaft model

Validate rotor model with measurements of free-free modes (room Temp)

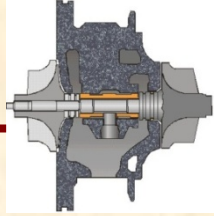


Rotor: 6Y gram
SFRB: Y gram

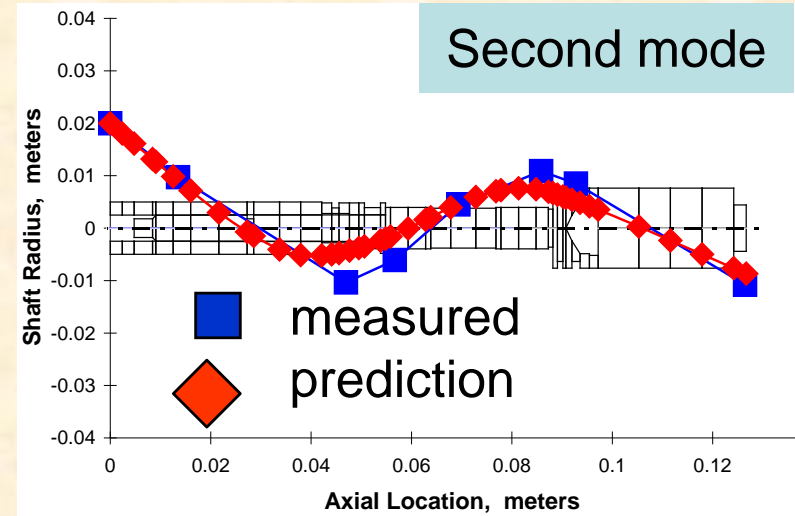
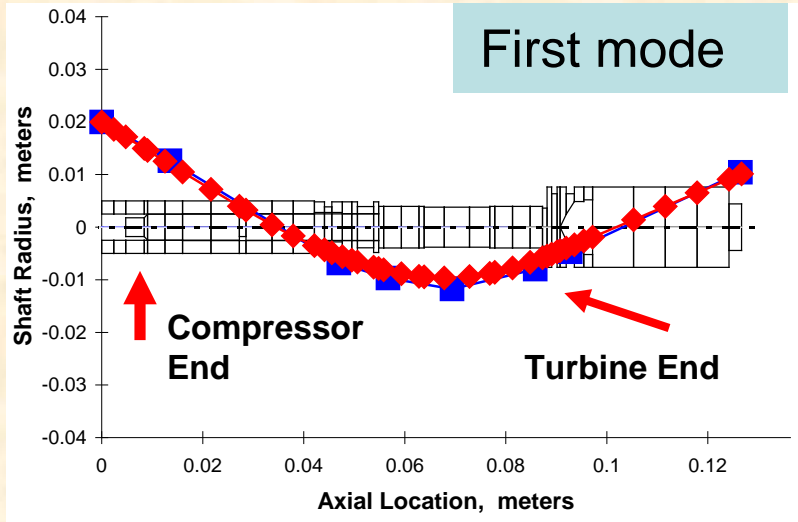
Static weight load distribution
Compressor Side: Z
Turbine Side: 5Z



Validate rotor model



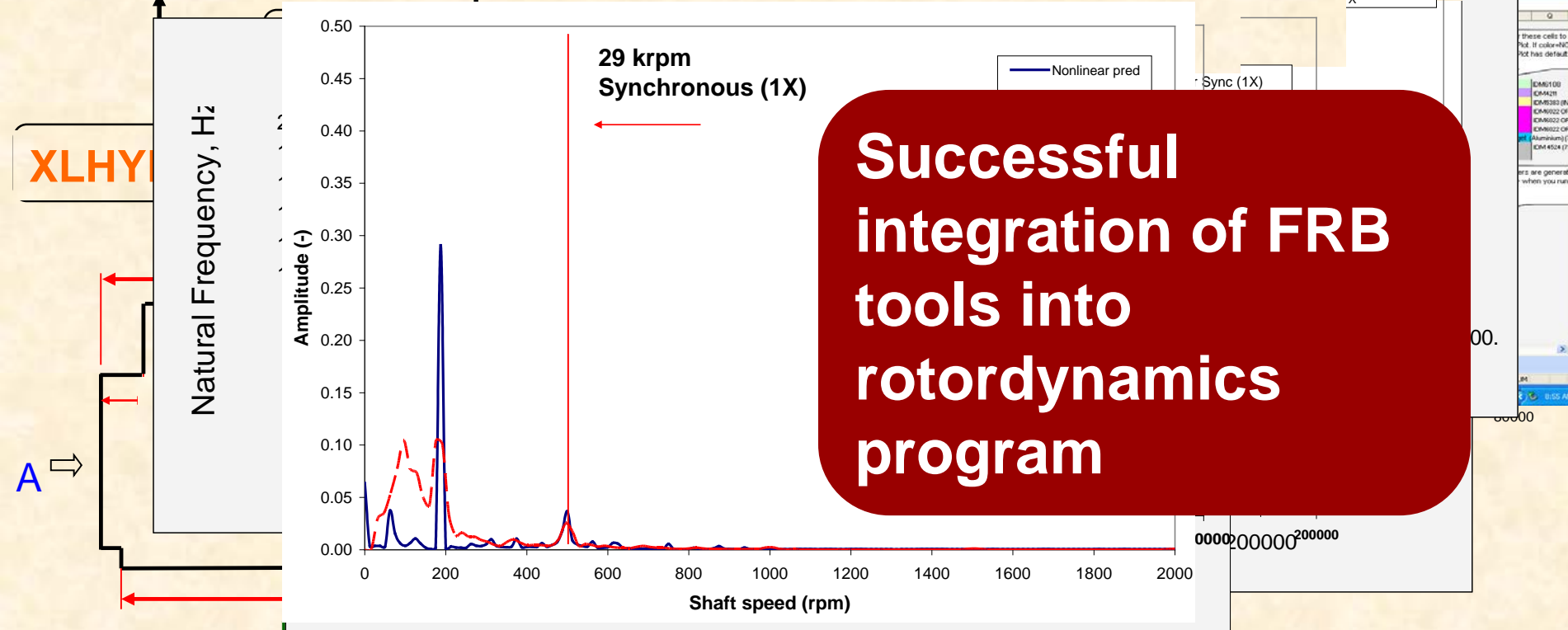
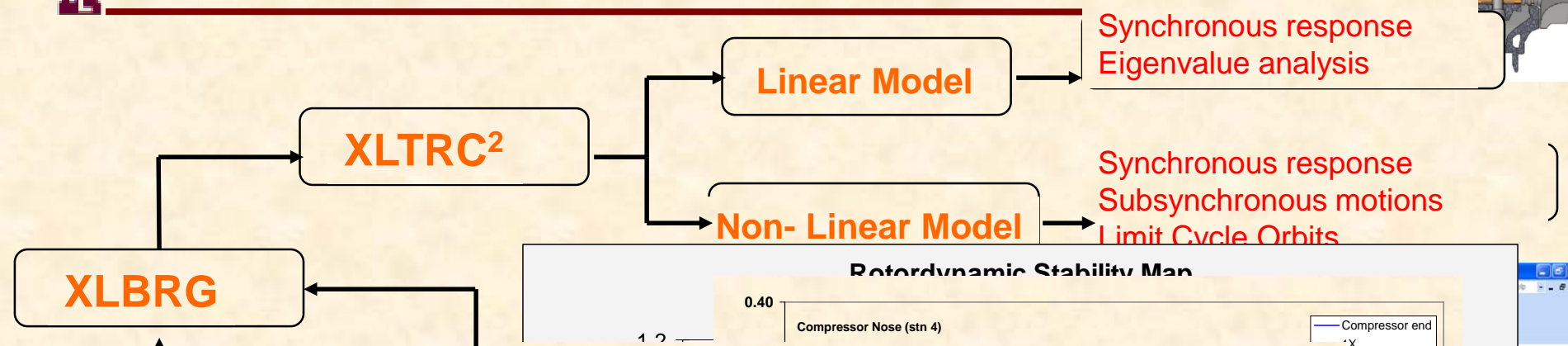
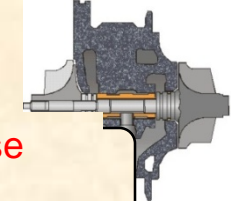
Free-free natural frequency & shapes



	measured	Predicted	% diff
	KHz	KHz	-
First	1.799	1.823	1.3
Second	4.938	4.559	7.7

Measured and predicted free-free natural frequencies and mode shapes agree: **rotor model validation**

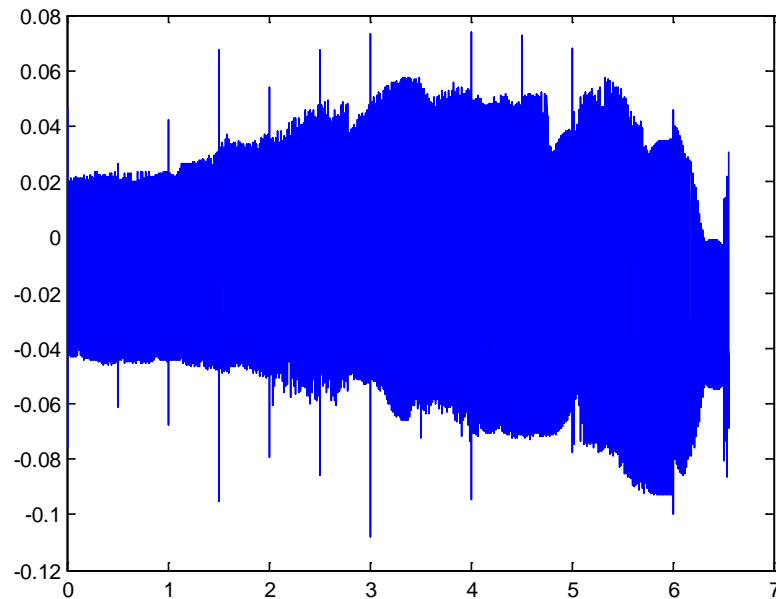
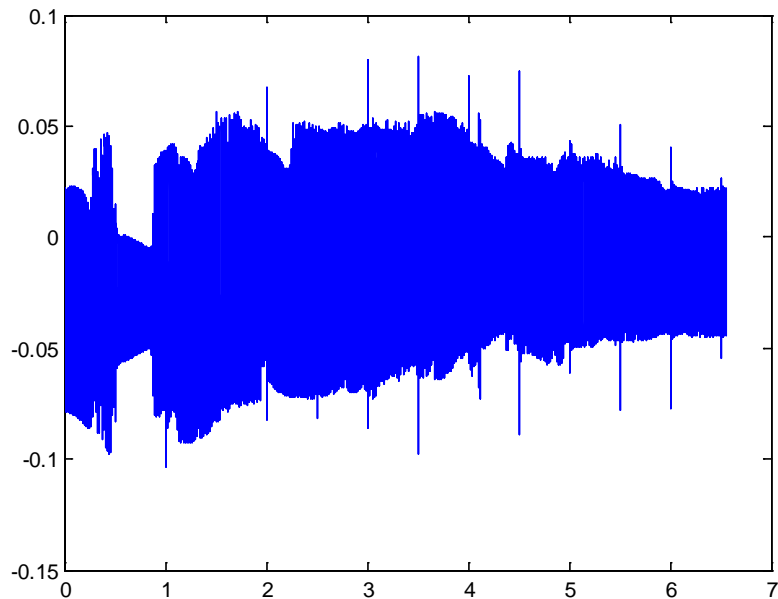
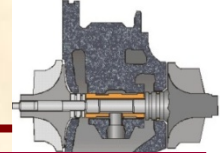
XLTRC² & XLBRG interfacing



Successful integration of FRB tools into rotordynamics program



NL predictions: typical responses



18 krpm

240 krpm 240 krpm

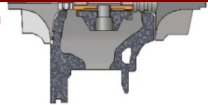
18 krpm



Predictions of TC shaft motion response – displacement versus time: rotor acceleration & deceleration



NL predictions: **analyses in frequency domain**



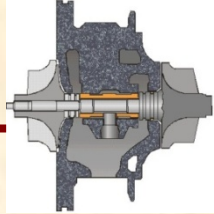
Important:

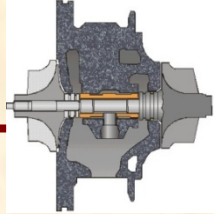
Massive amounts of time domain data rarely show any value (do not add knowledge nor establish firm design rules not even rules of thumb)

Analysis stresses on frequency domain analysis to build waterfalls, find total motion and synchronous motions, filtering of major whirl frequencies to determine effect on rotor elastic motions, calculation of forces transmitted to casing and rotor.



TC shaft motions virtual tool





Test data vs. predictions

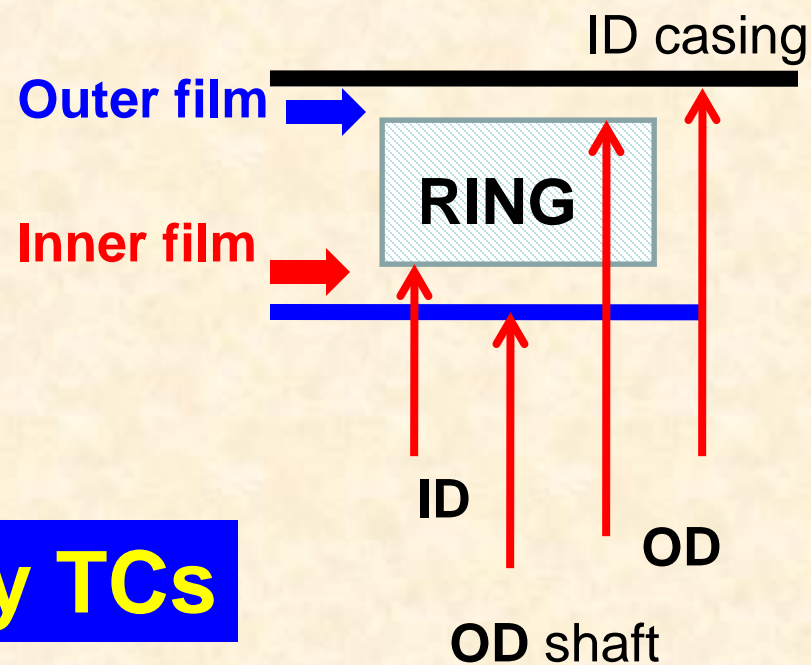
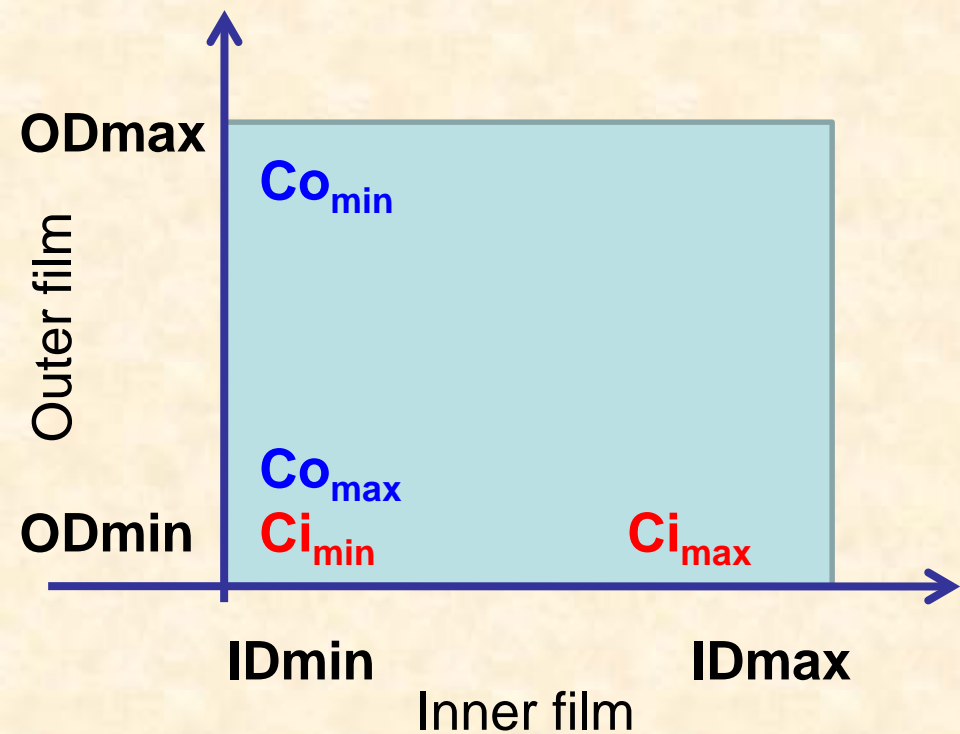
Validations

If successful,

- a) Ready tool for **PRODUCTION**
- b) Demonstrate savings
- c) Install tool at all TC core engineering centers



Variations in manufactured RING dimensions

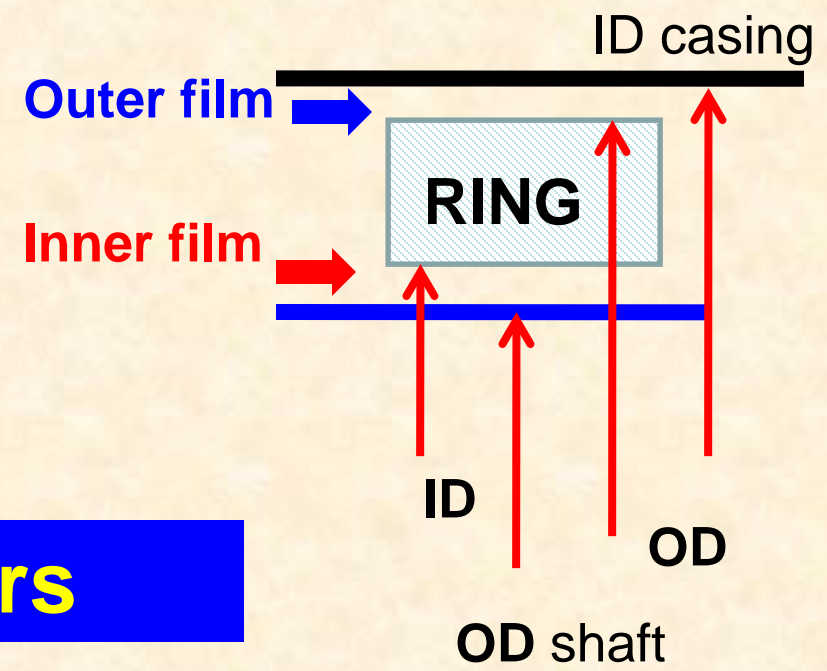
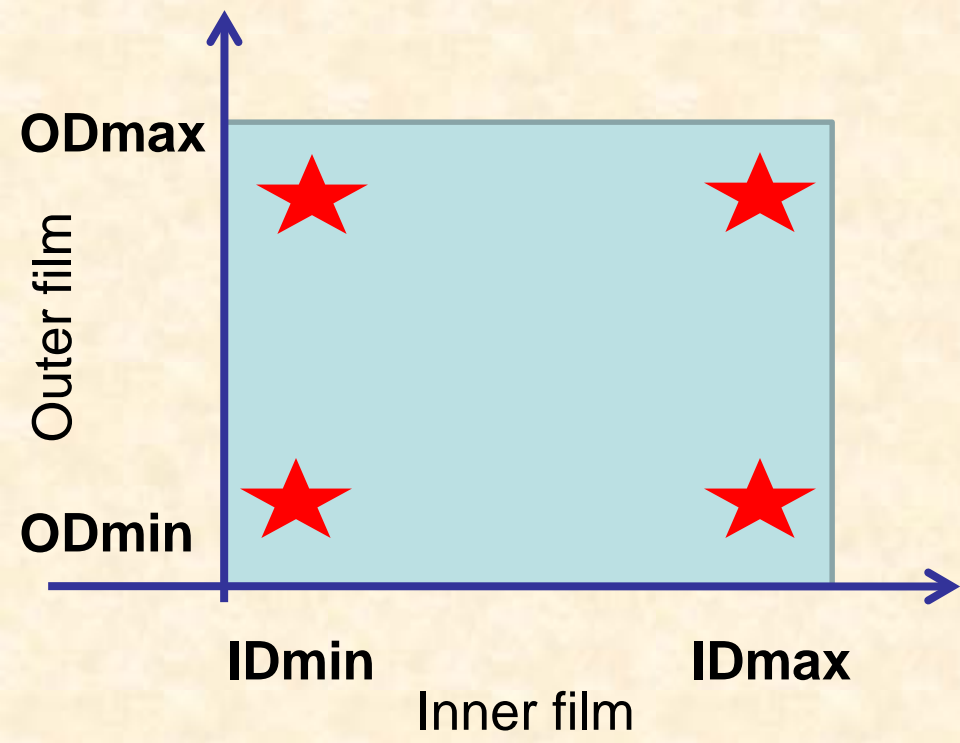


Costly procedure to qualify TCs

Four corners clearance limits



Costly TC qualification certification



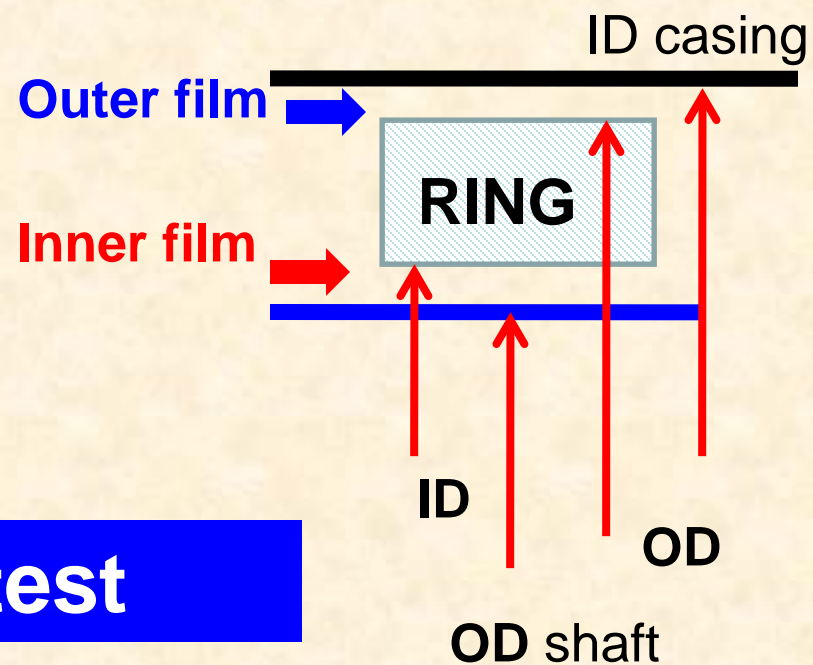
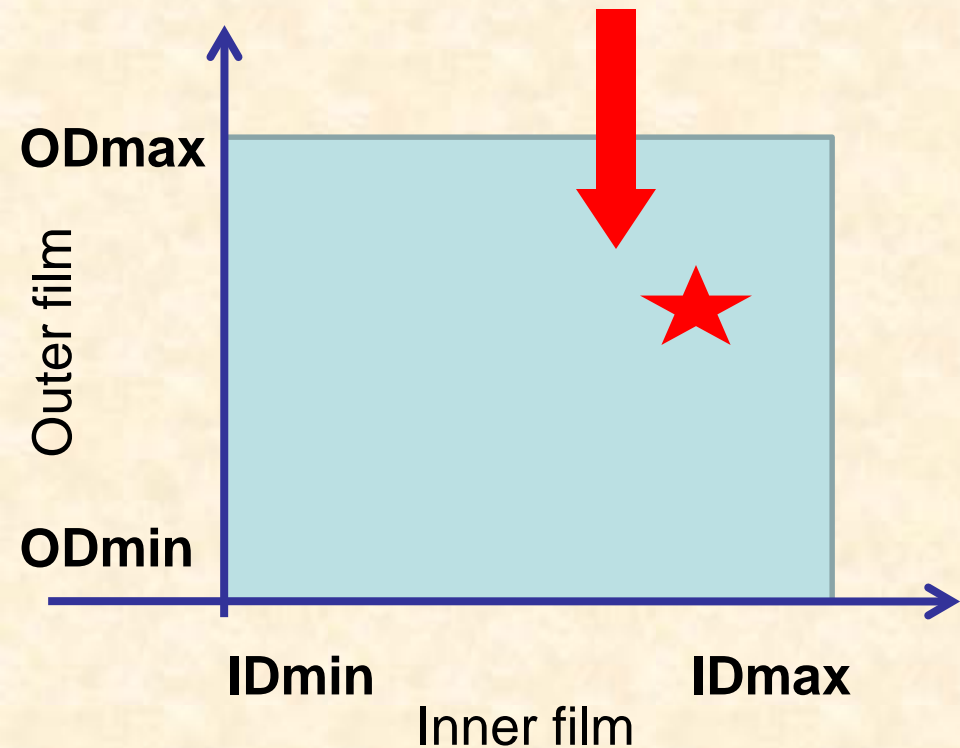
Past: NHS tests at 4 corners



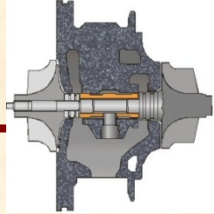
Savings in TC qualification certification



Determined from Virtual Tool



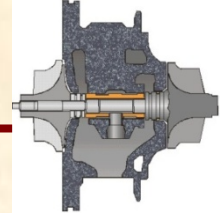
Current: One (or no) NHS test



Validation: shaft motion for PV TC



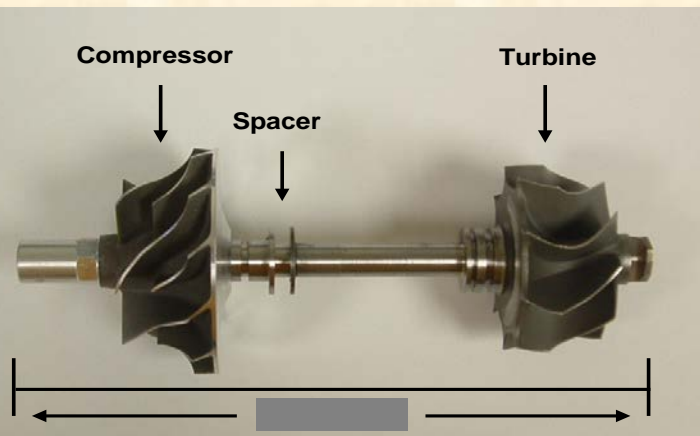
TC shaft motions virtual tool



TC rotor & bearing system

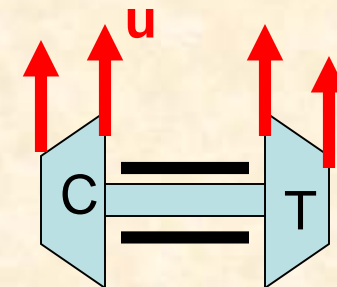
2 shaft model

Example: RBS
with Semi
Floating
Bearing

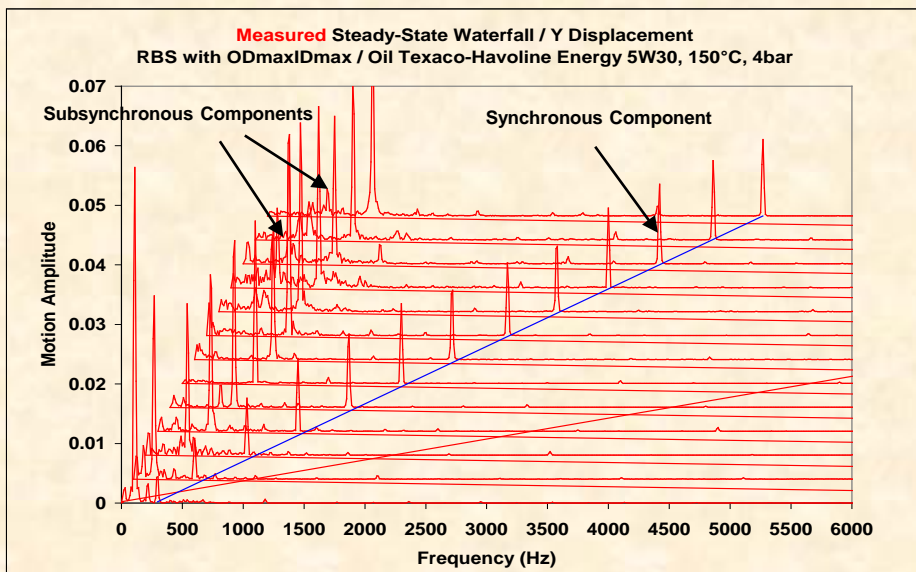
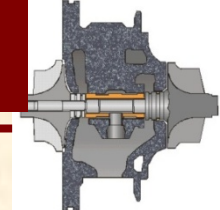


shaft speed **18 - 240 krpm**

Oil **5W-30**, 100 C inlet temperature, feed pressure 2,4 bar



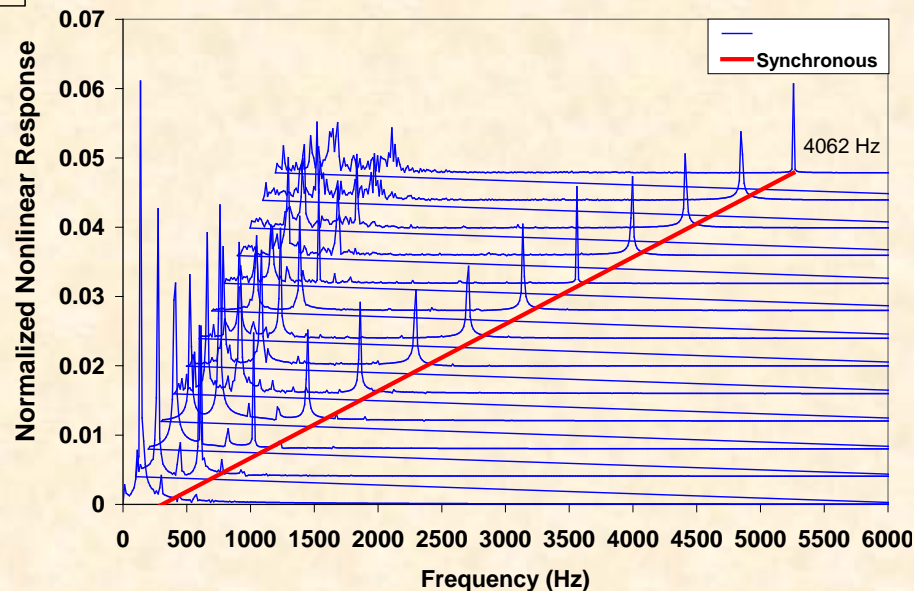
ODmax-IDmax - compare



Measured at compressor end

Predicted at compressor end

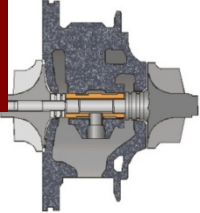
ODmaxIDmax, Oil 5W30, Inlet Temp. = 150°C, Inlet Pressure = 4bar



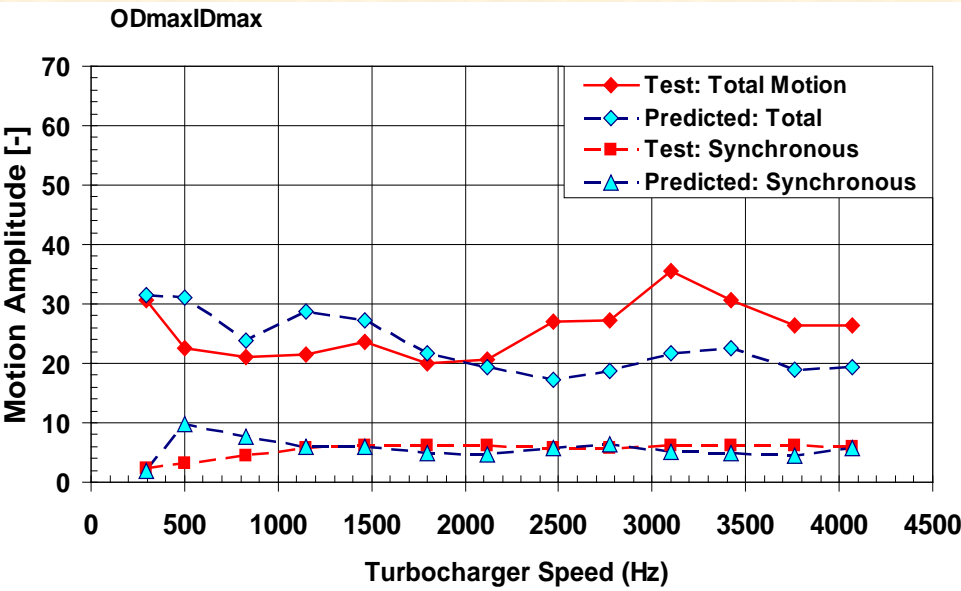
WATERFALLs of SHAFT MOTION

ASME DETC2007-34136

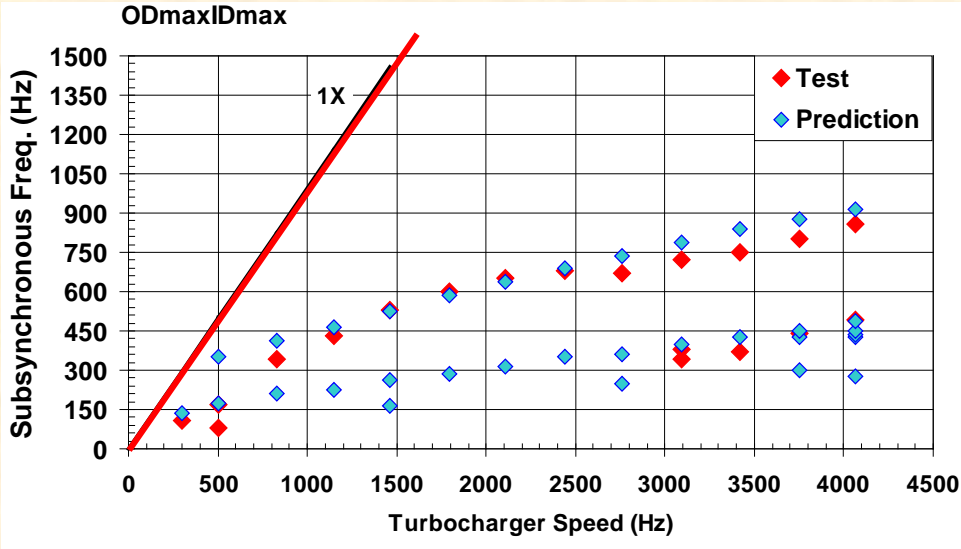
ODmax-IDmax - compare



Total motion & 1X motion

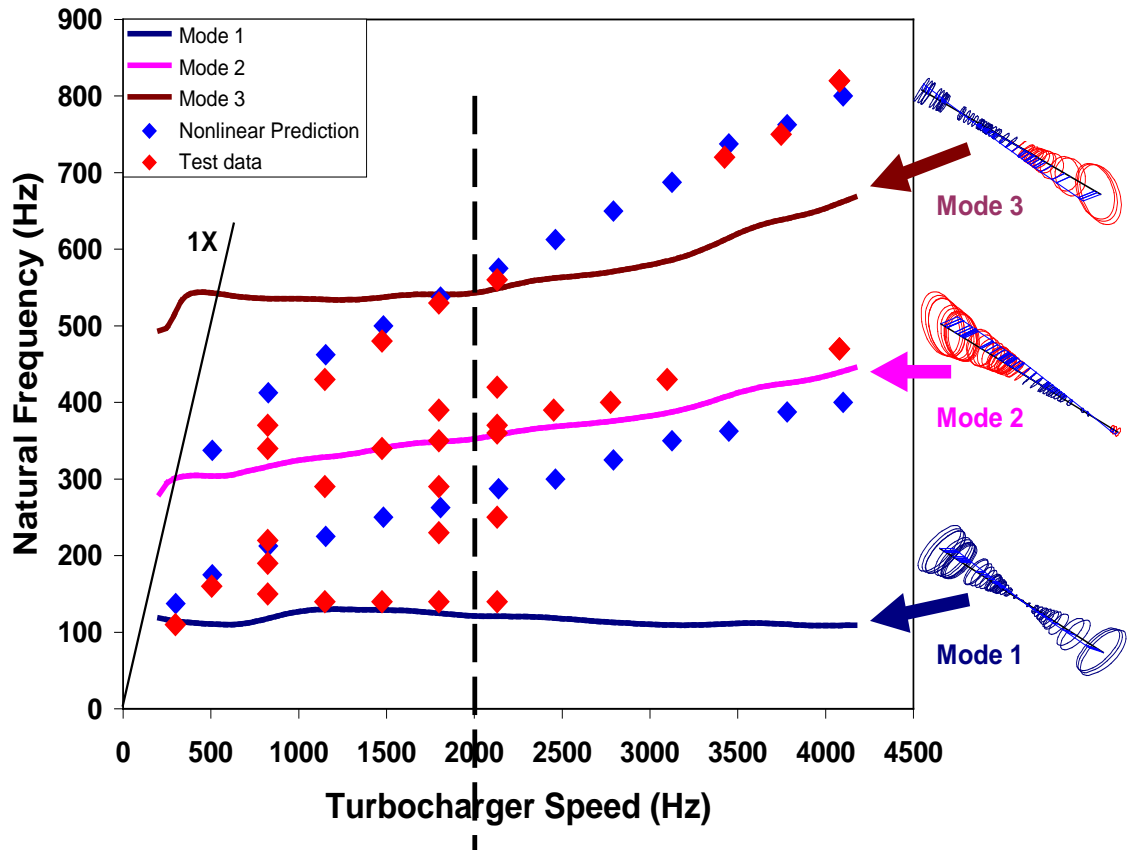
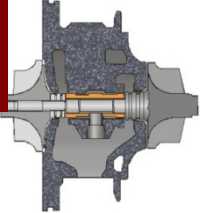


Whirl frequency





ODmin-IDmax - compare



Cylindrical - Deformed Mode Shape

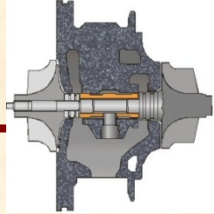
Compressor - End Ring Mode

Conical Mode Shape

Nonlinear predictions reproduce test data – Linear eigenvalue analysis is limited in accuracy

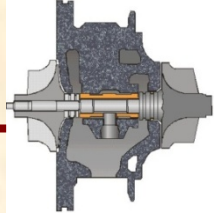


TC shaft motions virtual tool





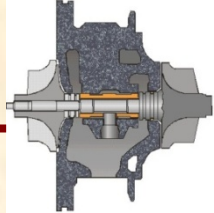
TC shaft motions virtual tool



Validation: shaft motion for CV TC



TC shaft motions virtual tool



TC rotor & bearing system

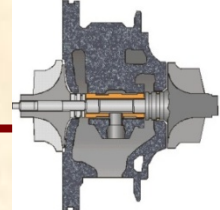
3 shaft model



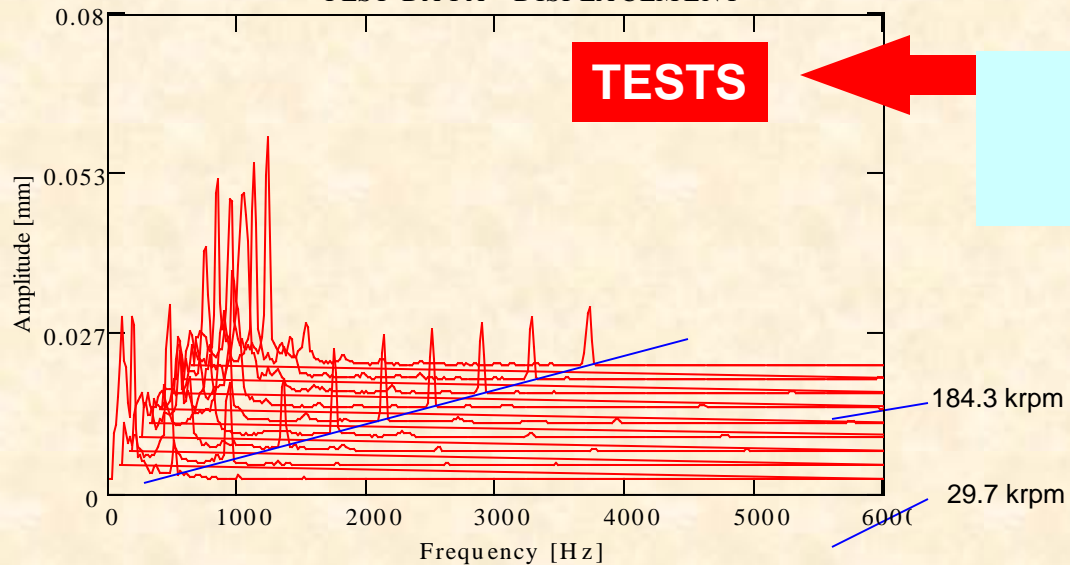
shaft speeds 30 - 180 krpm
Oil 0W-30, 92 C inlet temperature, feed pressure 4 bar



Validation CV TC



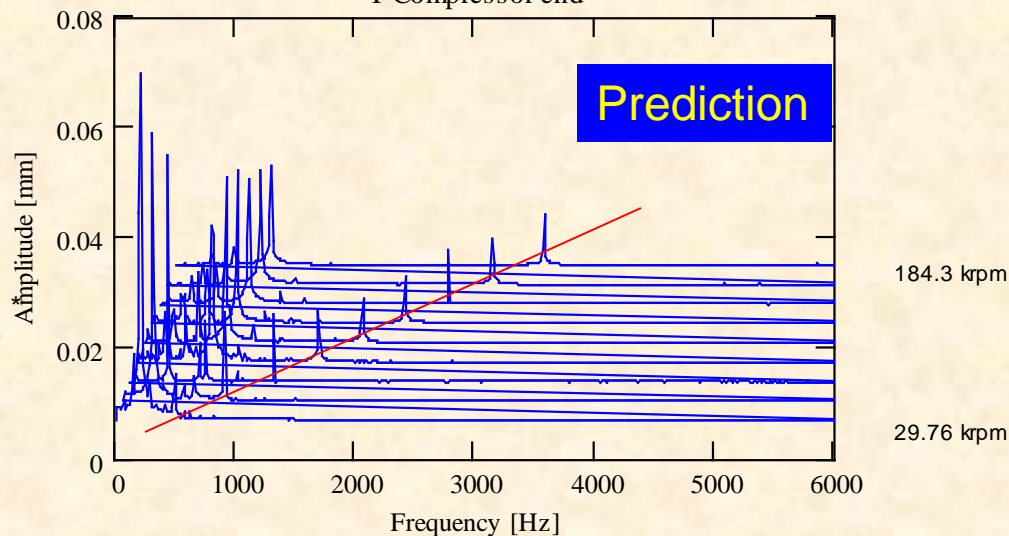
TEST DATA - DISPLACEMENT



Test data shows broad bands in sub synchronous frequency regions. Whirl motions persist at all speeds.

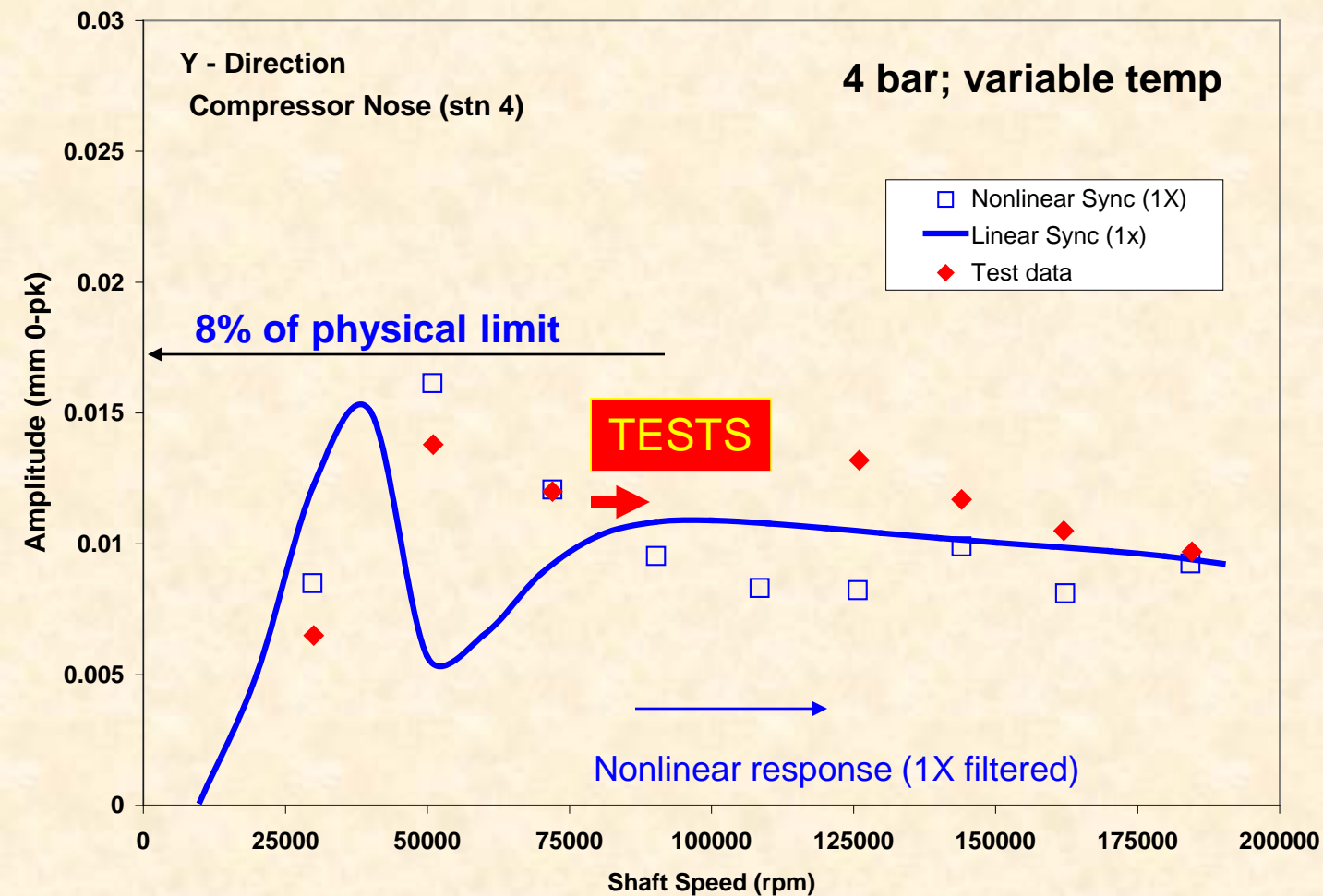
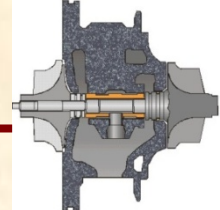
Predictions show sub synchronous frequencies to 184 krpm. More severe than test data at low shaft speeds.

Y-Compressor end





TC shaft motions virtual tool

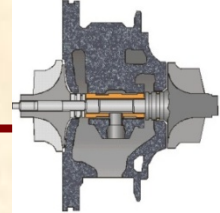


Nonlinear response predictions (1X filtered) compares best with test data at low shaft speeds

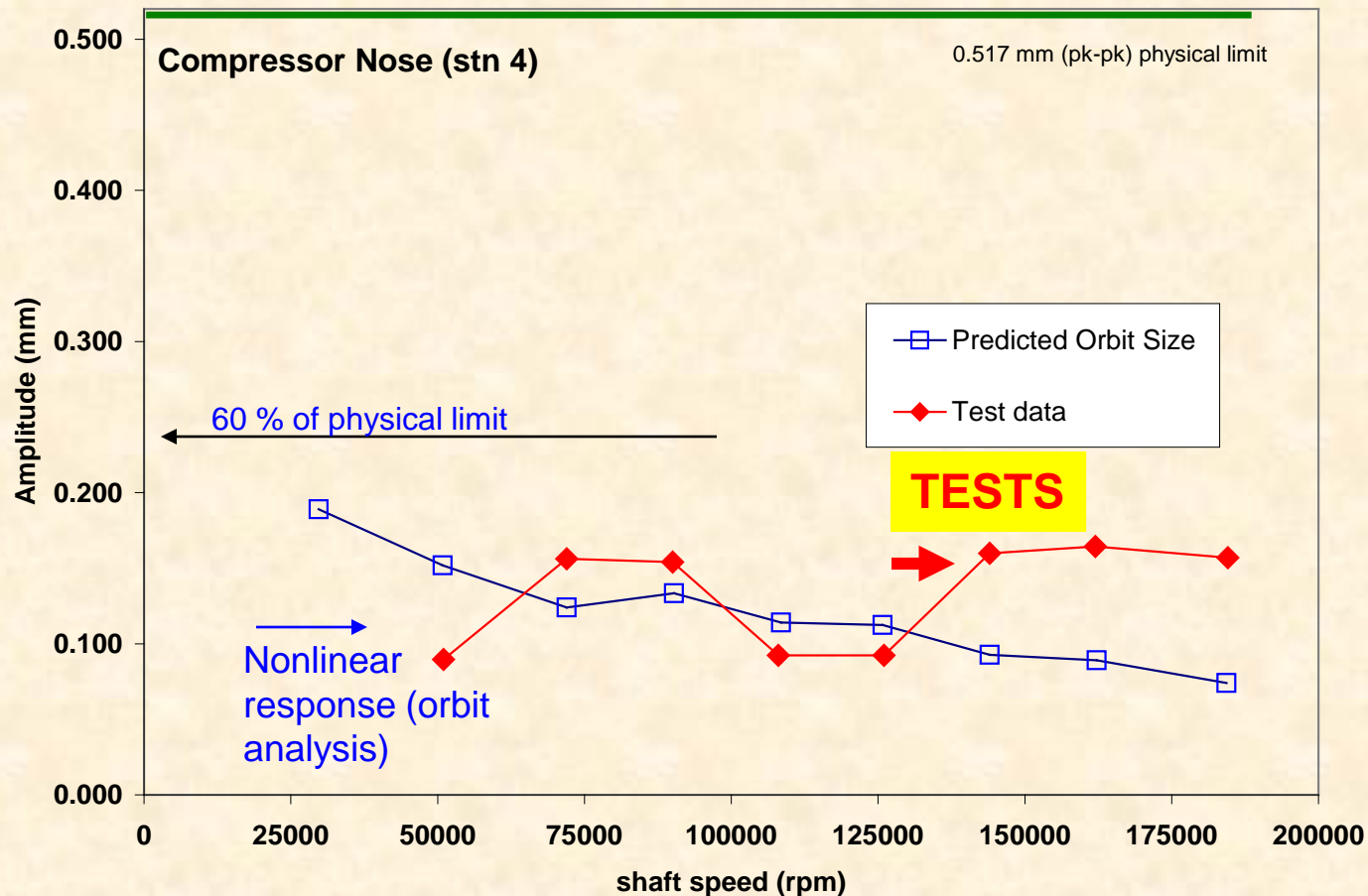
Imbalance response (linear and nonlinear) vs test data



TC shaft motions virtual tool



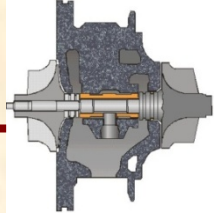
4 bar; variable temp



Good correlation with test data, in particular at mid shaft speed range (70-130 kprm).

Test data & predictions show persistent sub sync motions

Total Motion: test data and predictions

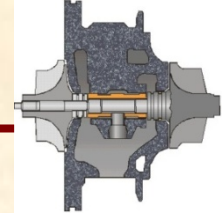


Validation: engine induced excitations

ASME GT 2009-59108

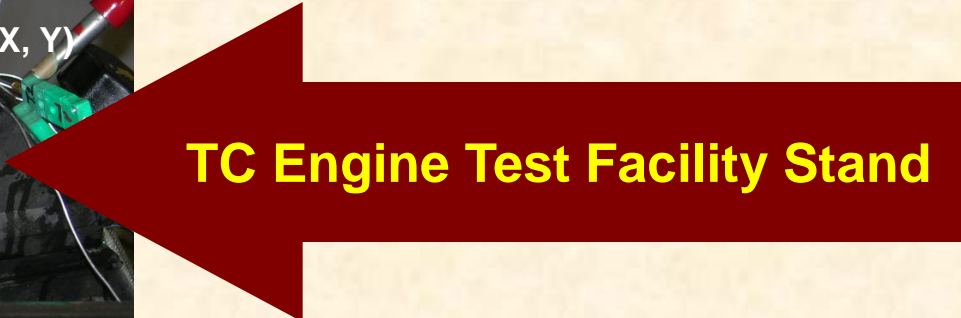
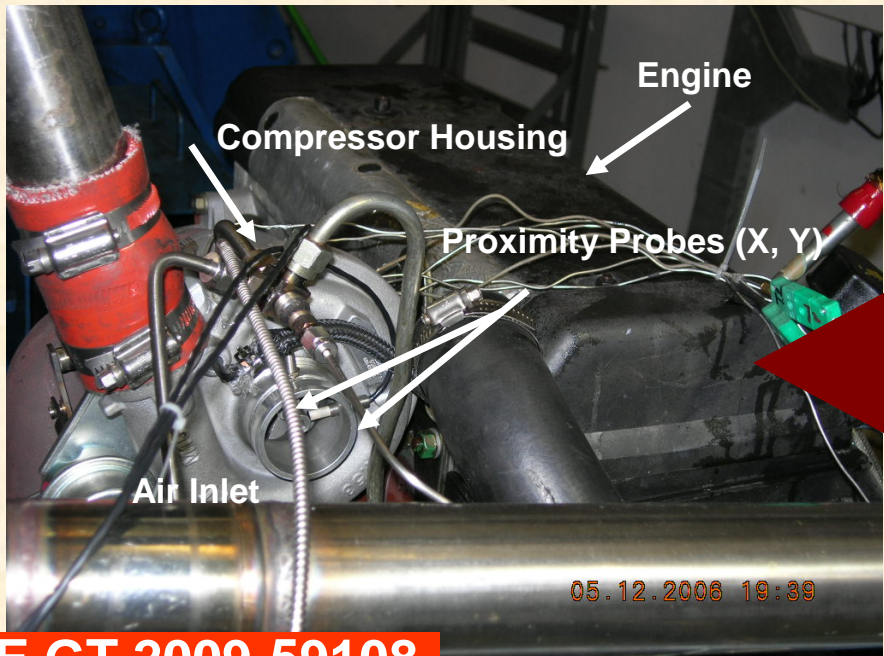


IC engine induced excitations



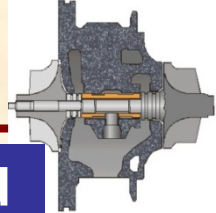
Operating conditions from test data:

- TC speed ranges from **48 krpm** – **158 krpm**
- Engine speed ranges from **1,000 rpm** – **3,600 rpm**
- **25%, 50%, 100%** of full engine load
- Nominal oil feed pressure & temperature: **2 bar, 100°C**





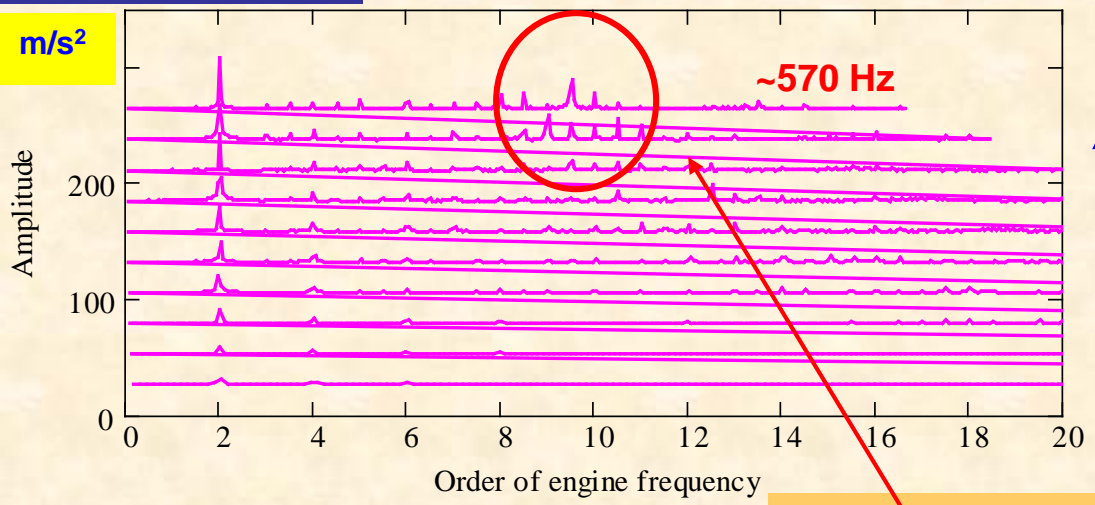
IC engine induced excitations



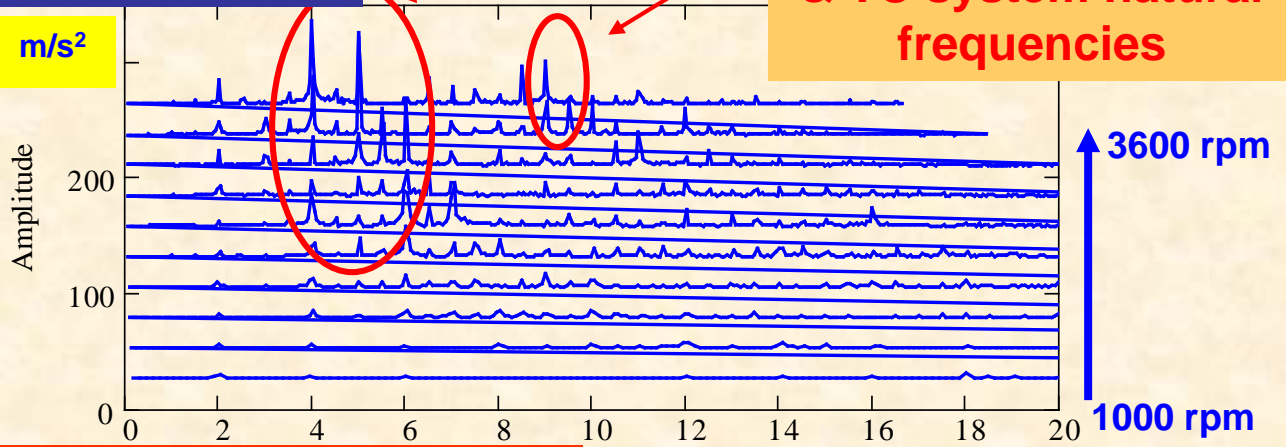
TC housing acceleration analysis

100% engine load

Center Housing



Comp. Housing

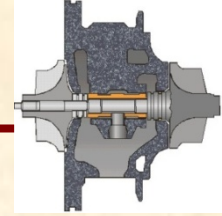


2, 4, and 6 times engine (e) main frequency contribute significantly

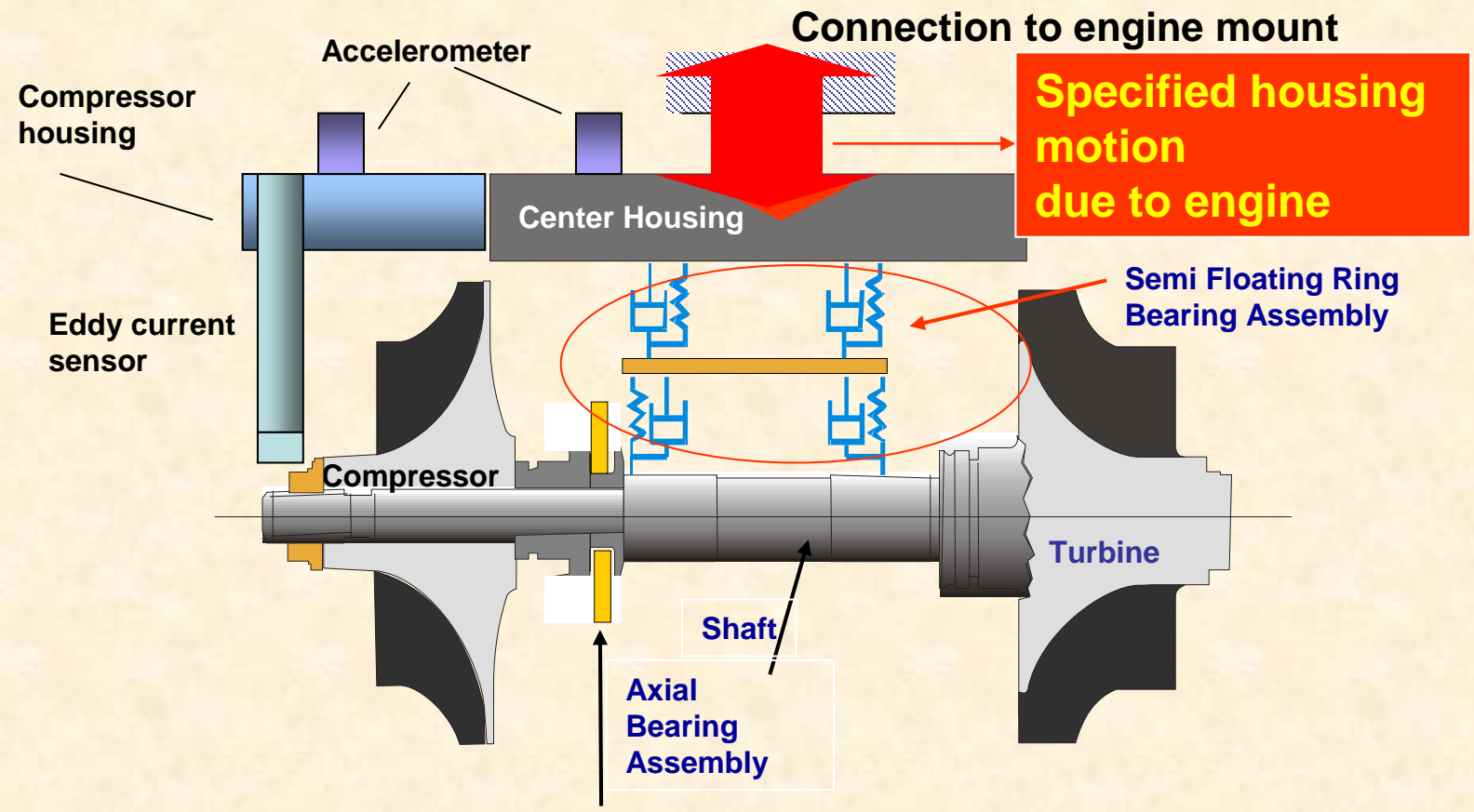
1e order frequency does not appear



IC engine induced excitations

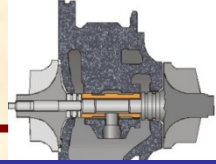


Housing accelerations into model

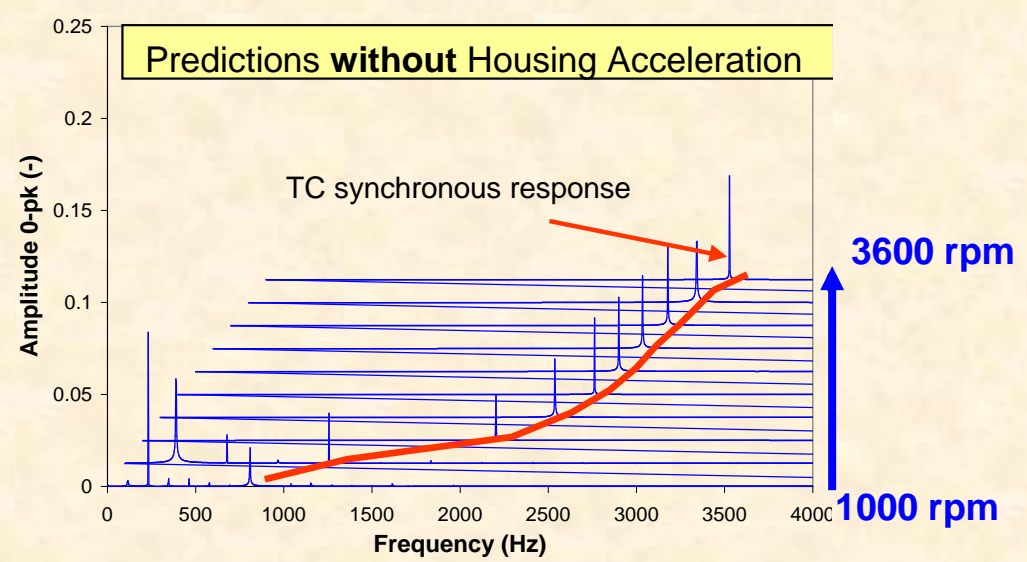
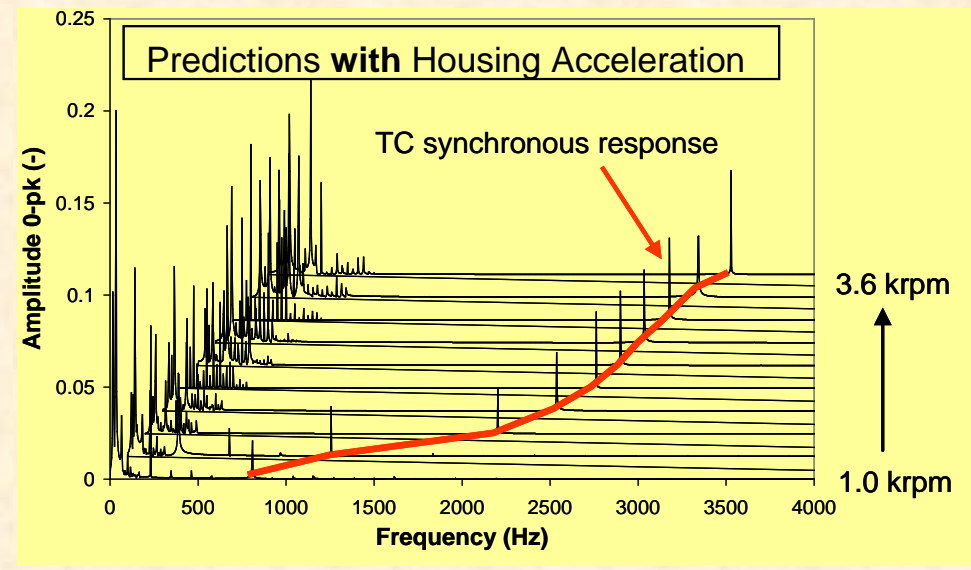
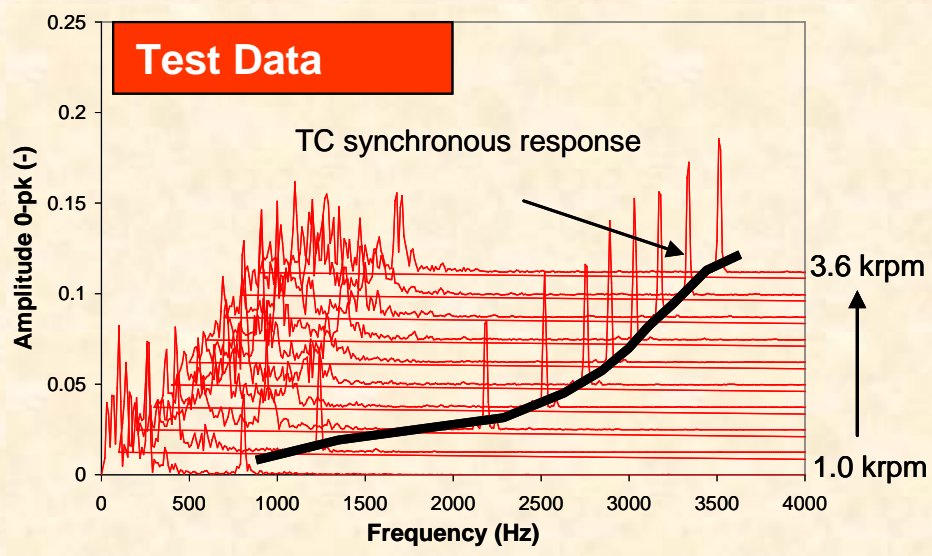




IC engine induced excitations



Waterfalls of shaft motion at compressor end 100% engine load

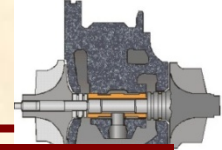


Housing accelerations induce broad range, low frequency whirl motions

Test data shows broad frequency response at low frequencies (engine speeds)

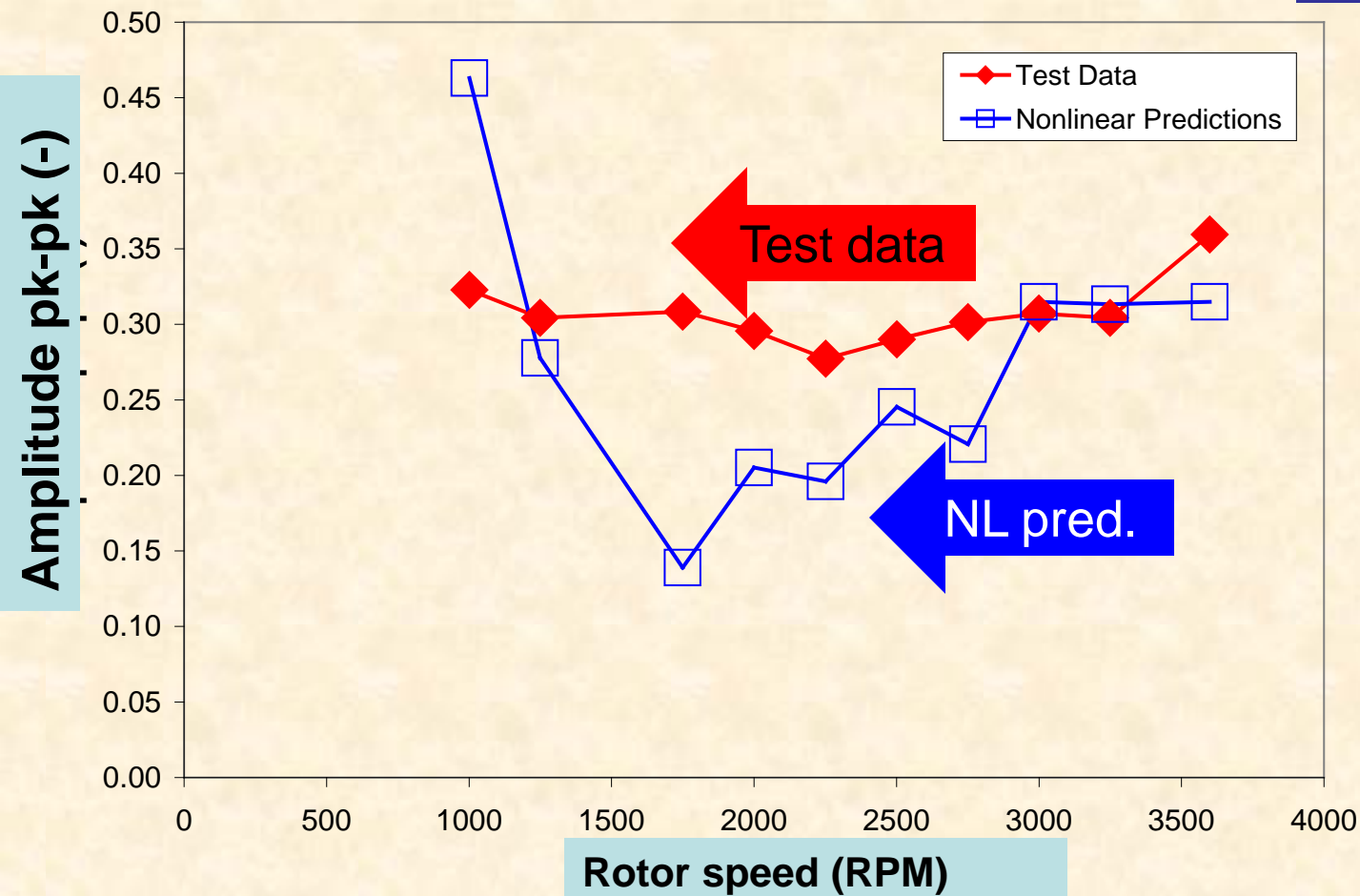


IC engine induced excitations



Total shaft motion at compressor end (amplitude)

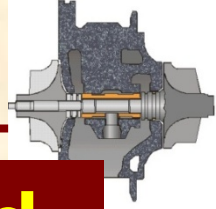
100% engine load



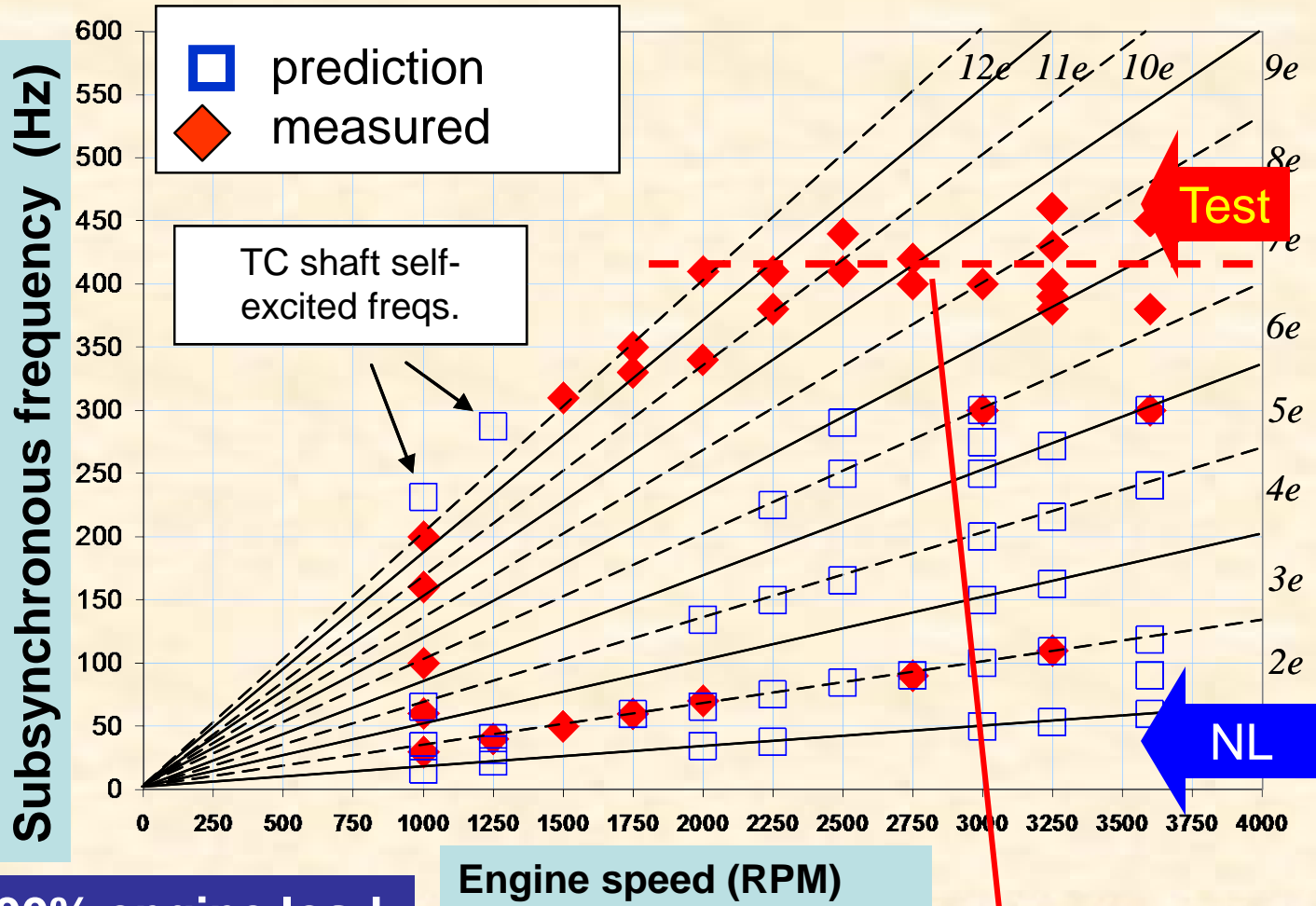
Good correlation with test data for all shaft speeds



IC engine induced excitations



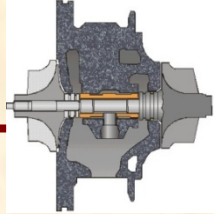
Subsynchronous freq. vs. IC engine speed



Subsynchron. freqs. are multiples of IC engine frequency

Higher engine order frequencies not predicted

TC manifold nat freq.

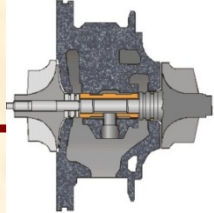


Validation: noise generation & frequency jump

IFTtoMM 2010

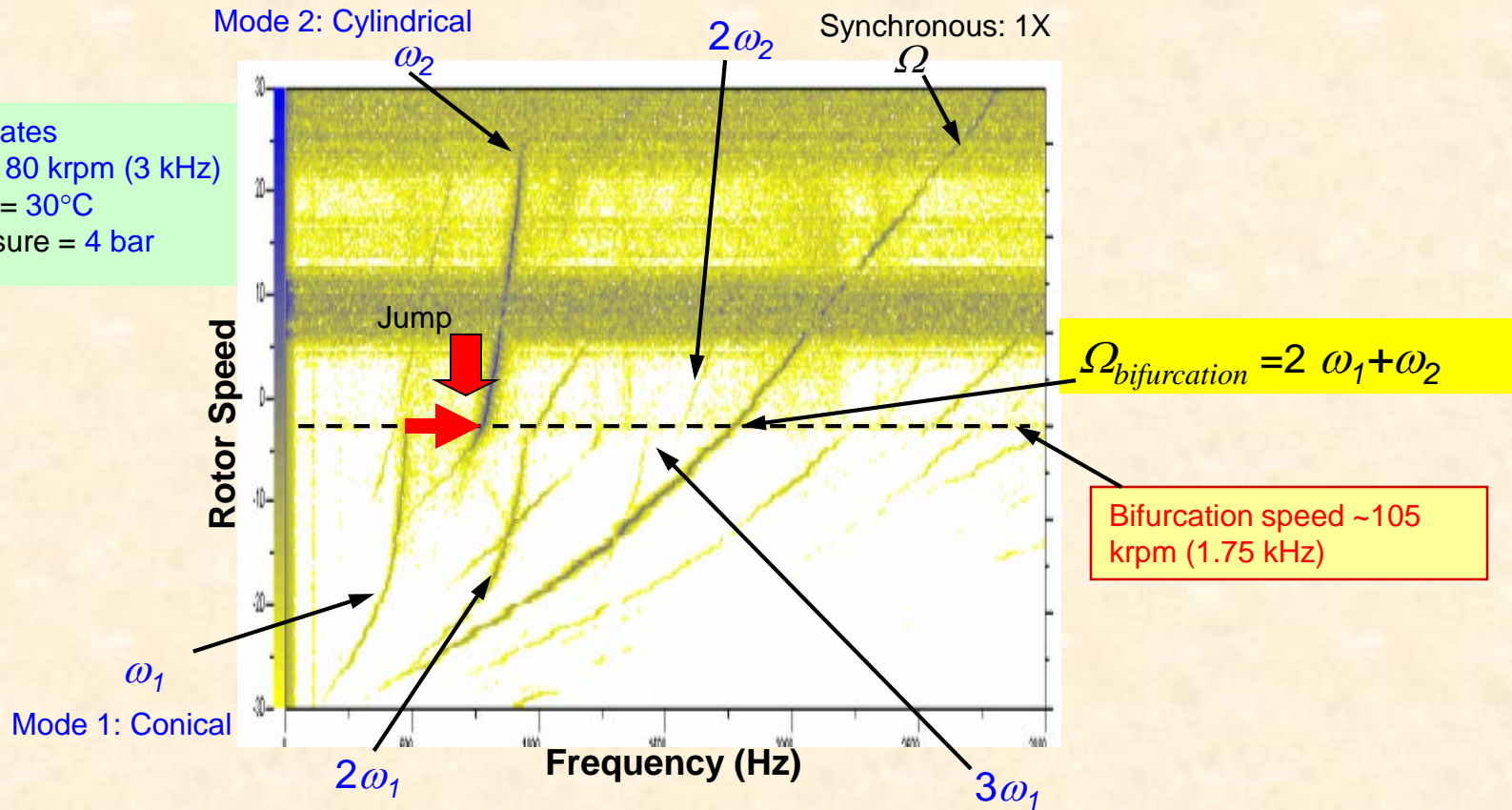


Frequency jumps: test data



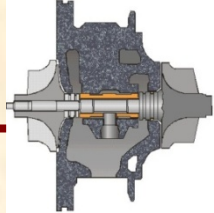
center housing acceleration (test data)

Jump from 1st to 2nd whirl frequency increases noise

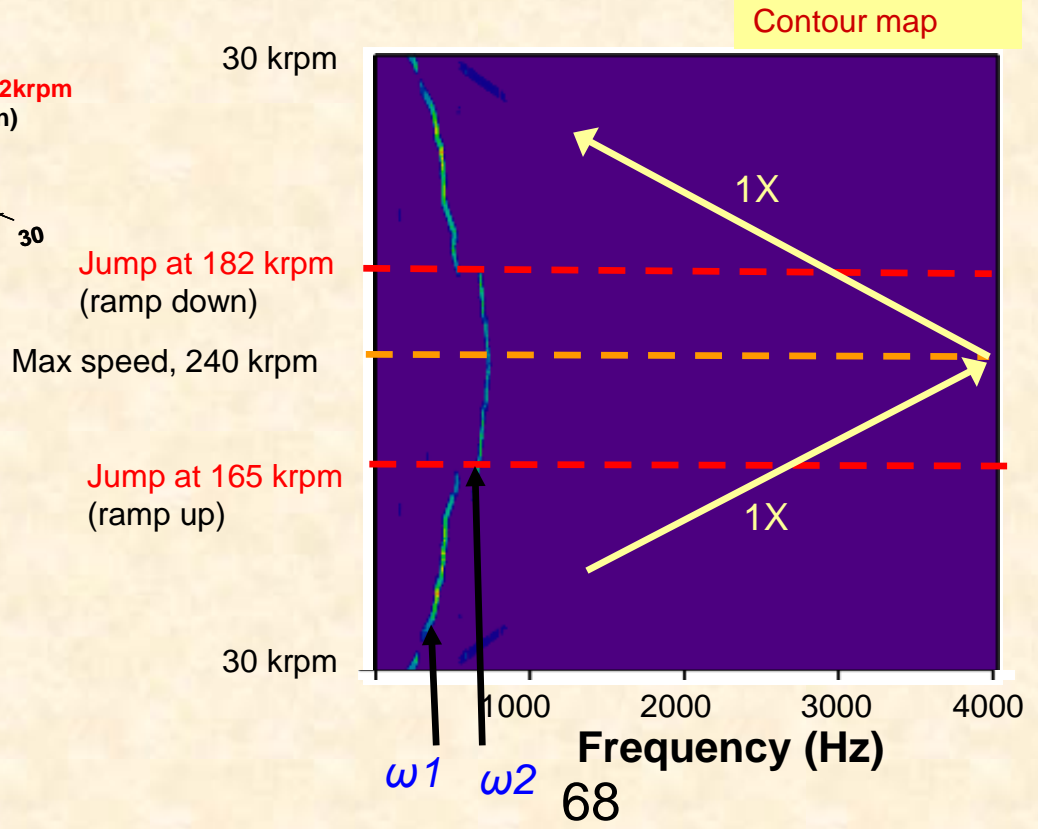
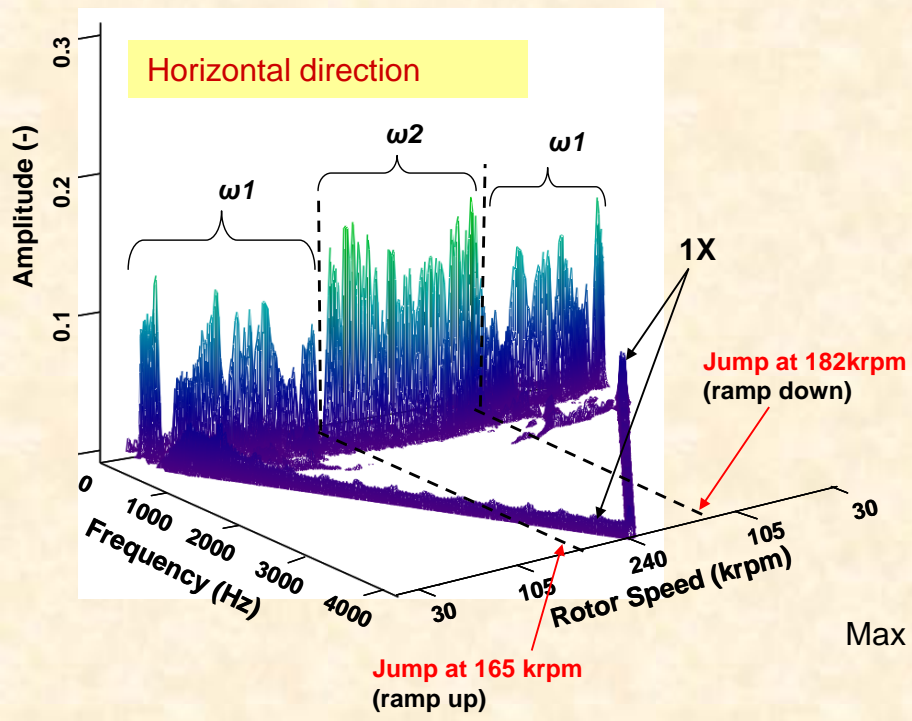


Objective: study bearing parameters and rotor characteristics affecting frequency jump

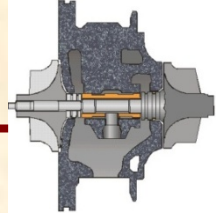
NL predictions: frequency jumps



Waterfalls of shaft motion (compressor end)

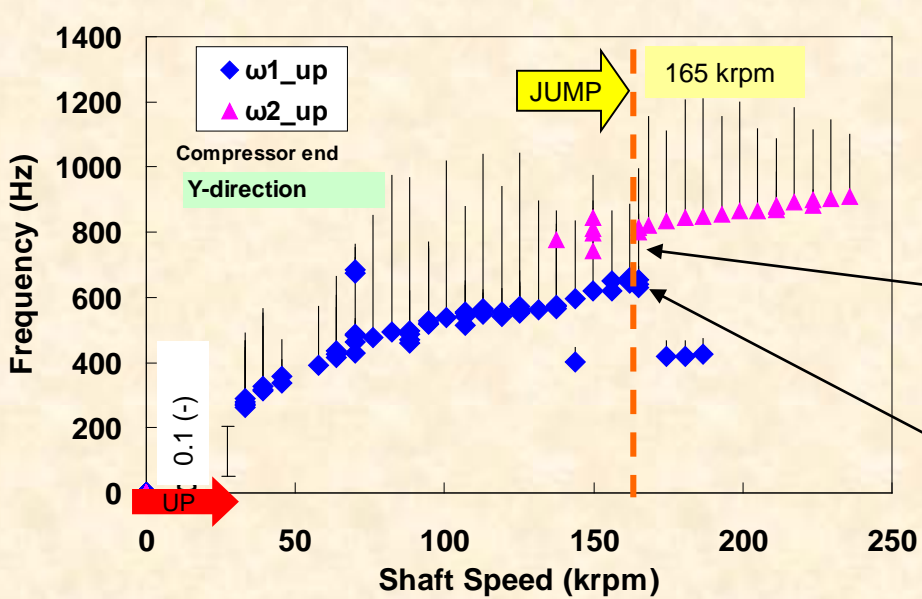


NL predictions: frequency jumps



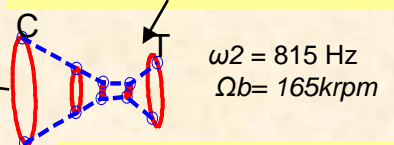
Rotor subsynchronous frequency (and amplitude) versus shaft speed (compressor end)

Rotor accelerates

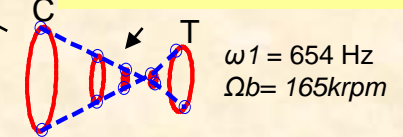


@ $\Omega_b = 165 \text{krpm}$ (2.75kHz)
 $5\omega_1 \sim 4\omega_2$
 $3\omega_1 + \omega_2 \sim \Omega_b$

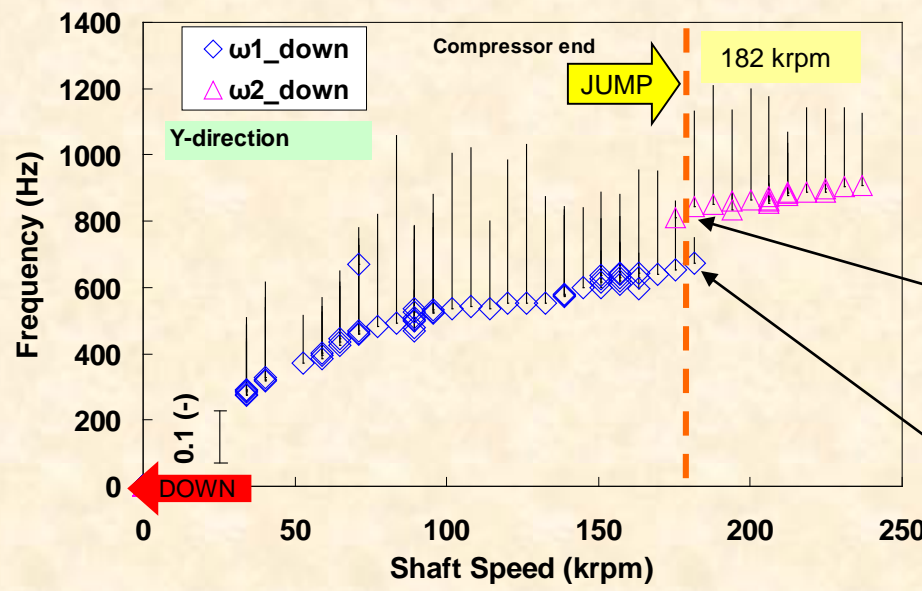
Cylindrical bending rotor filtered whirling mode



Conical rotor filtered whirling mode

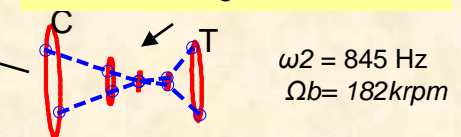


Rotor decelerates

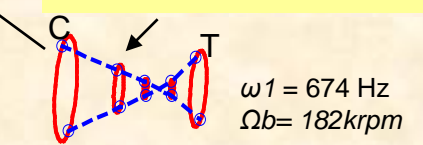


@ $\Omega_b = 182 \text{krpm}$ (~3kHz)
 $5\omega_1 \sim 4\omega_2$
 $2\omega_1 + 2\omega_2 \sim \Omega_b$

Cylindrical bending rotor filtered whirling mode



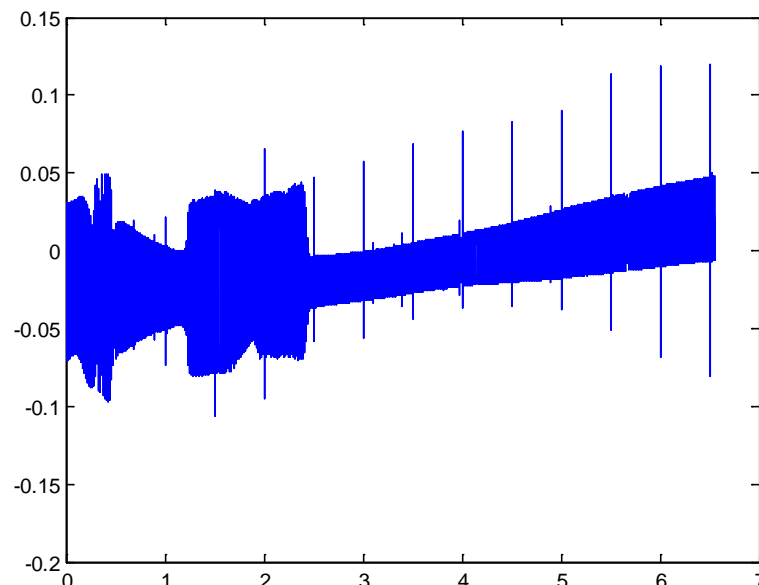
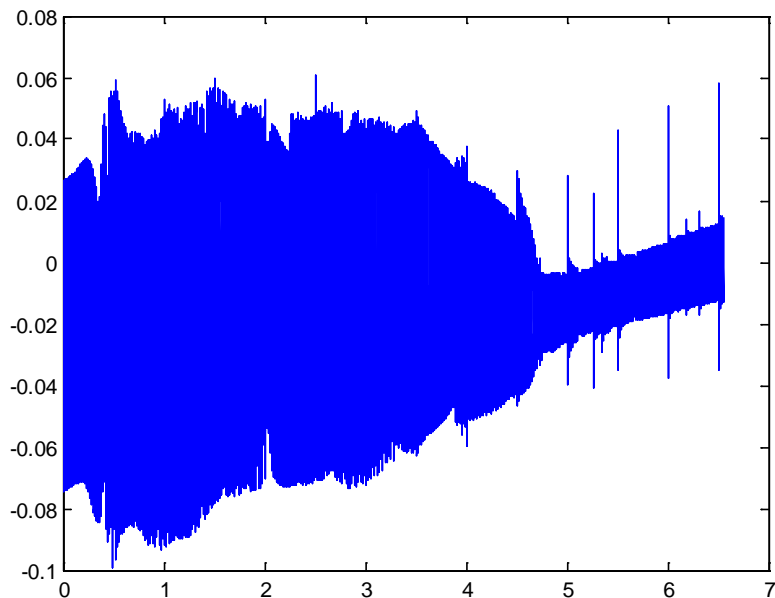
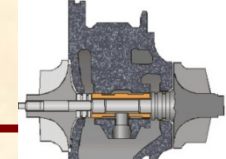
Conical rotor filtered whirling mode



IFTtoMM 2010



NL predictions: noise



18 krpm

240 krpm

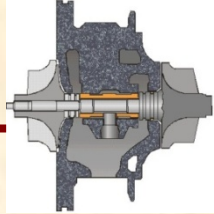
18 krpm

240 krpm

Predictions of TC shaft motion response – displacement versus time: rotor acceleration

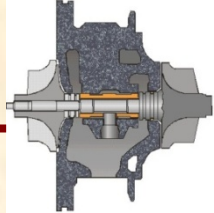


TC shaft motions virtual tool





TC shaft motions virtual tool



1. Tests SHOW dominance of SUB SYNCHRONOUS MOTIONS on rotordynamic response of PV TCs

TAMU & HTT

2. TOOL for prediction of fully floating and semi-floating ring bearing (SFRB) static and dynamic forced response is **ACCURATE**

XLBRG

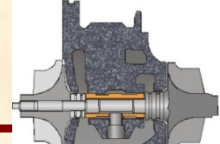
3. VIRTUAL TOOL: Seamless Integration of FRB and SFRB codes into nonlinear rotordynamics program

XLTRC2

Test vs.
predictions

Substantial savings in product development/prototype testing

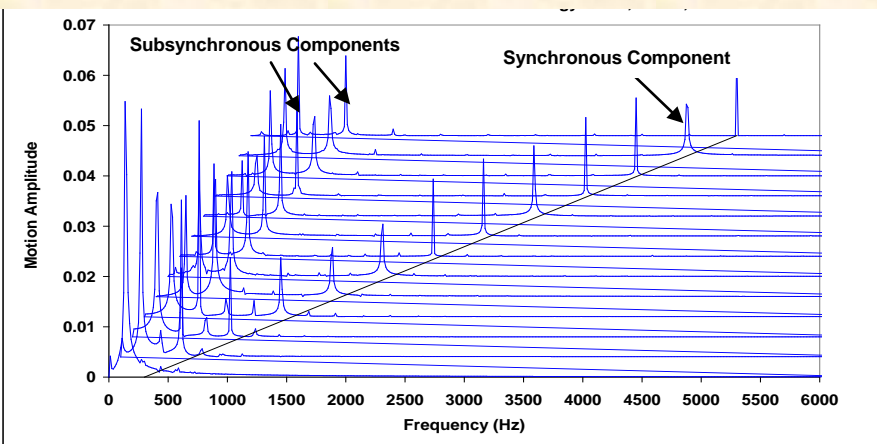
Major benefit to industry



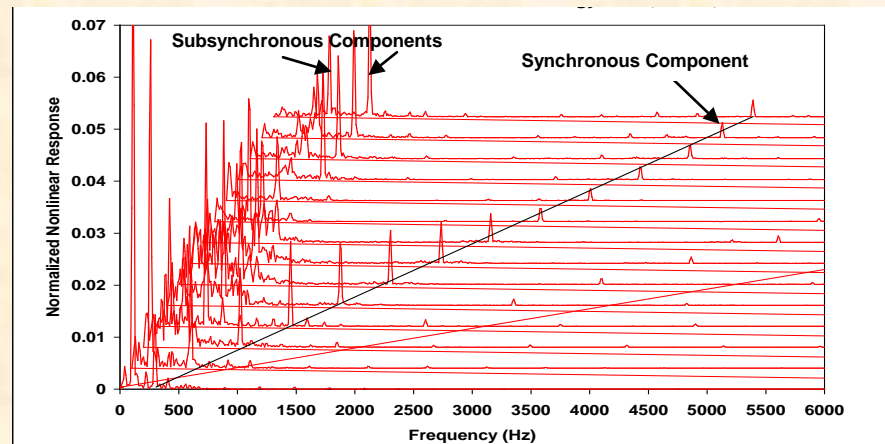
TAMU-HTT VIRTUAL TOOL for Turbocharger NL Shaft Motion Predictions

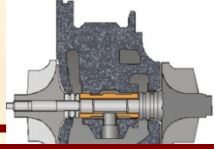
XLTRC²[®] & XLBRG[®] have a demonstrated 70% cycle time reduction in the development of new **CV TCs**. Since 2006, code aids to developing **PV TCs** with savings up to \$150k/year in qualification test time

Predicted shaft motion



Measured shaft motion



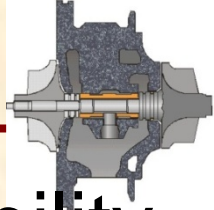


HTT 2011-12 Project

- Complete thermal analysis of FRBs and S-FRBs for TCs
- Prediction of thermal fields in entire TC system
- Quantification of power losses and prediction of bearing seizure & oil coking
- Analysis of frequency jump phenomena and multiple internal and combined resonances
- \$ 350 k (2 years)



Oil-less turbochargers



Driver: HT ceramic ICEs with improved reliability

Advantages:

+ TH efficiency, HT limited by materials only, less contamination

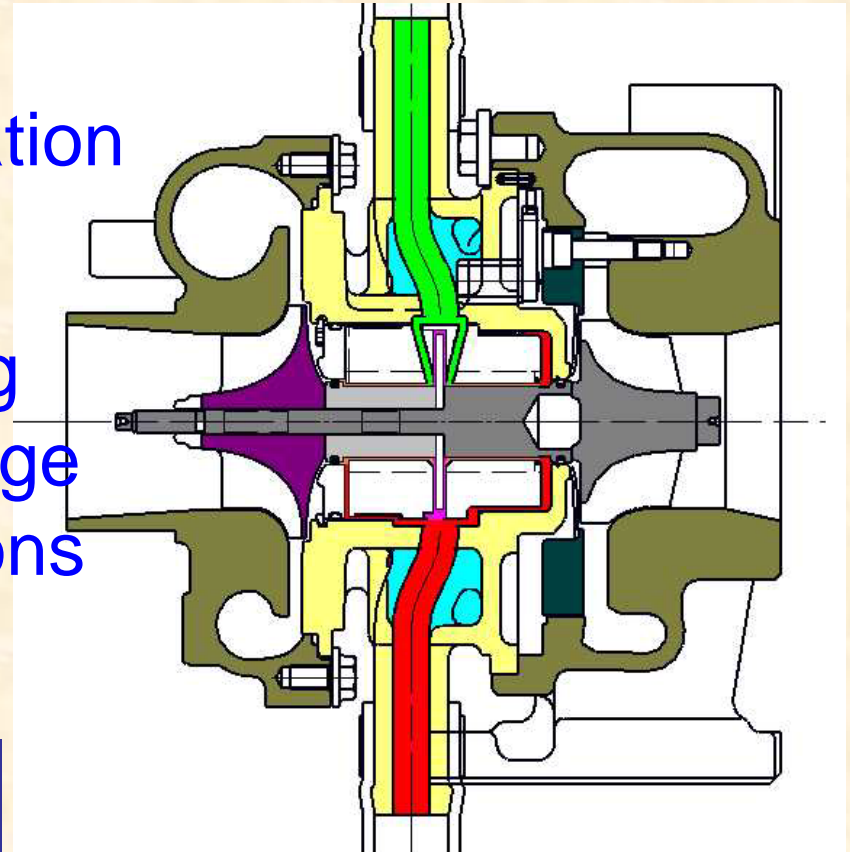
Disadvantages:

+ cost, more parts & balancing

Unknown performance for large dynamic loads & road conditions

Unknown thermal soaking

**Cheap solution sought:
metal wire mesh bearings!**





Other forces and issues

Thrust bearings:

Tools available

Issues: thermal & coupling to lateral RD in PV TCs

Aerodynamic forces:

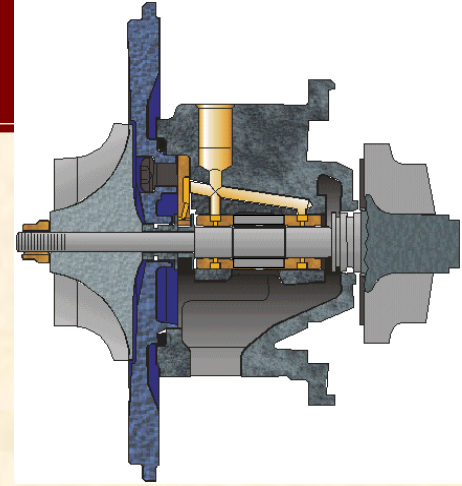
Tools available

Issue: At + high speeds, turbine develops a destabilizing force

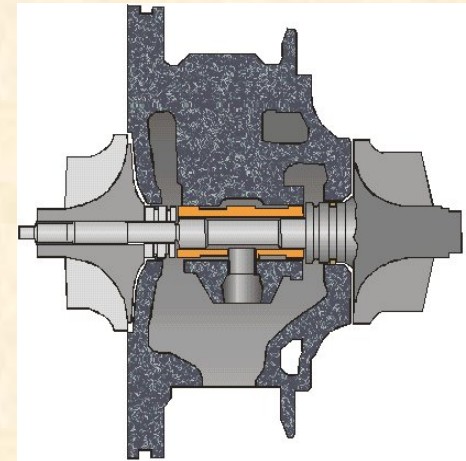
Piston ring seal:

Unknown forces.

Issue: oil coking locks ring



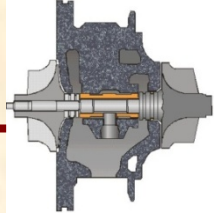
CV TC



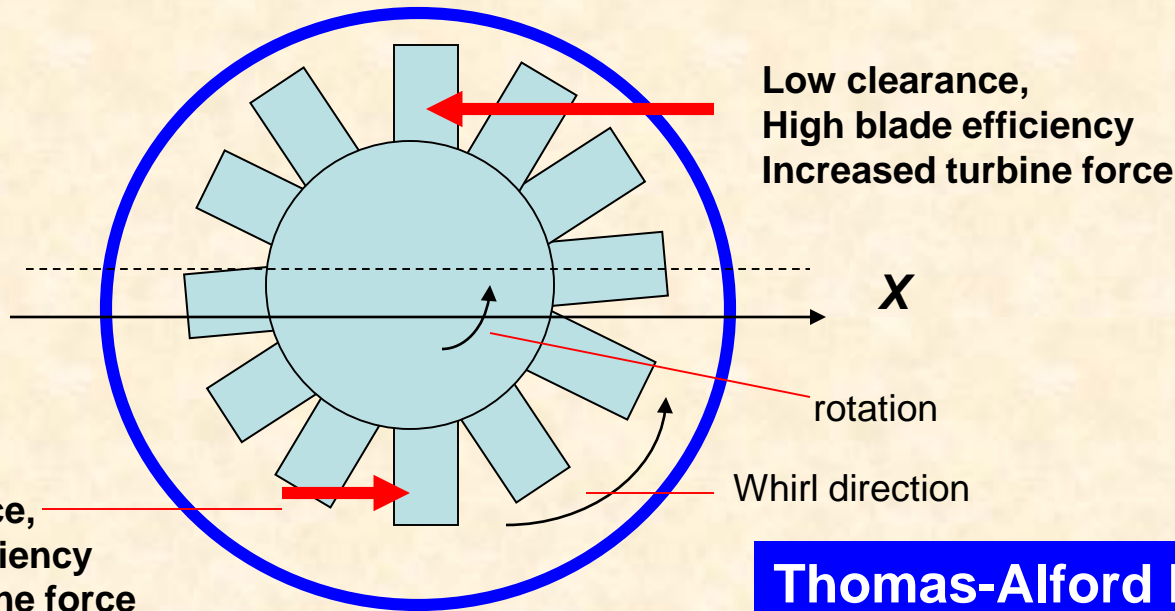
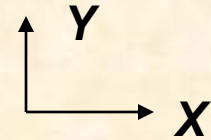
PV TC



Aerodynamic force in turbines



As rotor whirls, regions of low clearance improve efficiency of blades and generate a force (from torque)



Thomas-Alford Force Model

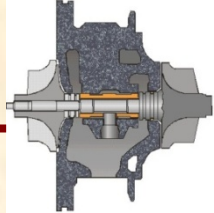
$$-F_X = K_{XY} \Delta Y, -F_Y = -K_{YX} \Delta X$$

$$K_{xy} = -K_{yx} = \frac{\beta T}{DH}$$

- T:** torque
- D:** tip diameter
- H:** blade height
- β:** efficiency parameter (empirical) =1-1.5

Tip Clearance Excitation Force

Review



Acknowledgments

**Honeywell Turbocharging Technologies
(2002-2011)**

TAMU Turbomachinery Laboratory

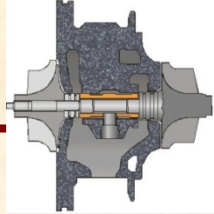
**Turbomachinery Research Consortium
(XLTRC²[®])**

**Luis San Andres
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<http://rotorlab.tamu.edu>**



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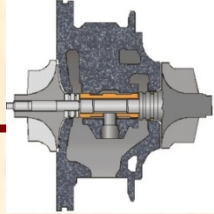


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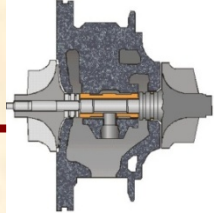


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	Over 80 proprietary monthly progress reports to sponsor (Honeywell Turbocharging Systems), 2002-2011.

References



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